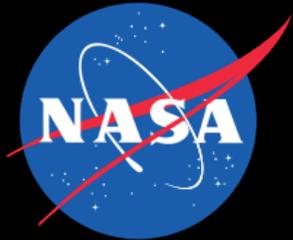


Implantation of Helium into JSC-1A Lunar Regolith Simulant for Volatile Extraction System Testing



Aaron D.S. Olson
G.L. Kulcinski, J.F. Santarius,
University of Wisconsin-Madison
Fusion Technology Institute



J.G. Mantovani
NASA Kennedy Space Center

7th SRR/PTMSS
Golden, CO, June 9th, 2016

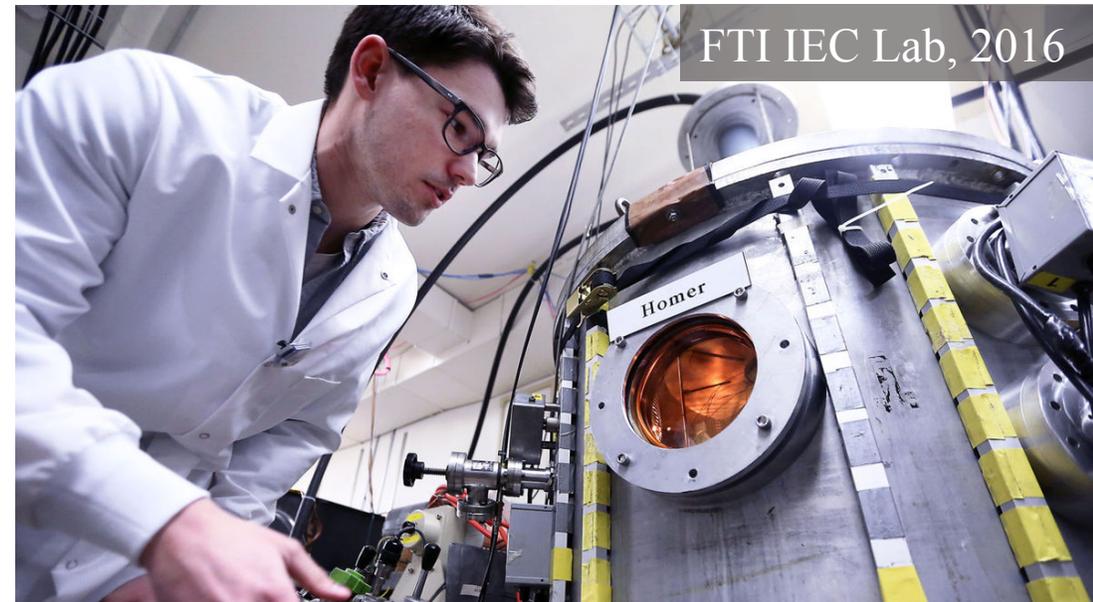


Outline

- Motivation and Context for the Presented Research
- Overview of Research to Demonstrate Lunar Volatiles Extraction
- Solar Wind Implanter (SWIM) Design and Operation
- Conclusion: Progress and Future Work

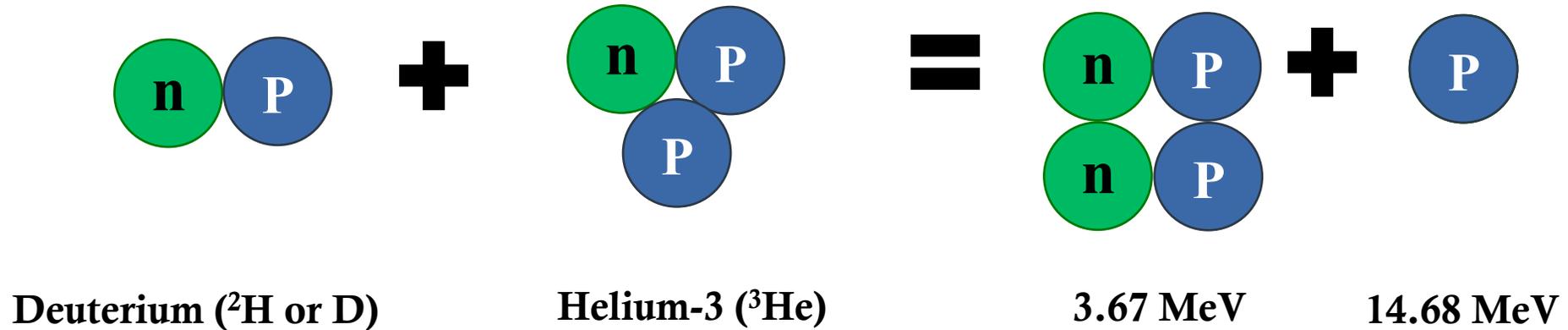
Fusion Technology Institute

- Large government programs: ITER, NIF
- Nearer term applications: neutron sources, medical isotopes
- Commercial fusion companies
- Resources for fusion (Lunar ^3He)



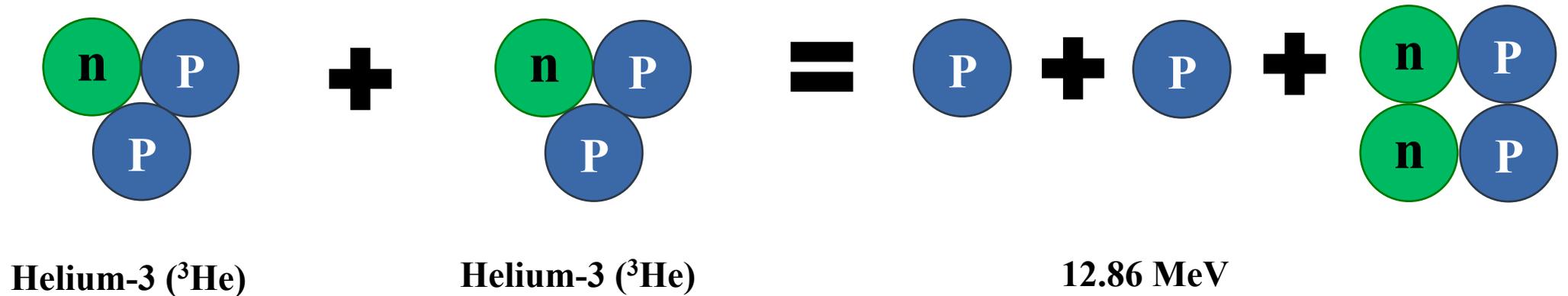
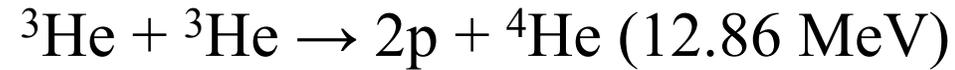
³He Fusion Reactors Could Produce Nuclear Power with Little to No Nuclear Waste

Deuterium-Helium-3



³He Fusion Reactors Could Produce Nuclear Power with Little to No Nuclear Waste

Helium-3 and Helium-3



^3He Fusion Reactors Could Produce Nuclear Power with Little to No Nuclear Waste

- No Greenhouse or Acid Gas Emissions
- Very High Energy Conversion Efficiencies ($>70\%$)
- Greatly Reduced Radiological Hazard Potential Compared to Fission Reactors ($<1/10,000$)
- No Possible Offsite Nuclear Fatalities in the Event of Worst Possible Accident
- No Proliferation of Weapons Grade Material

What is Required to Develop ^3He Fusion Reactors?

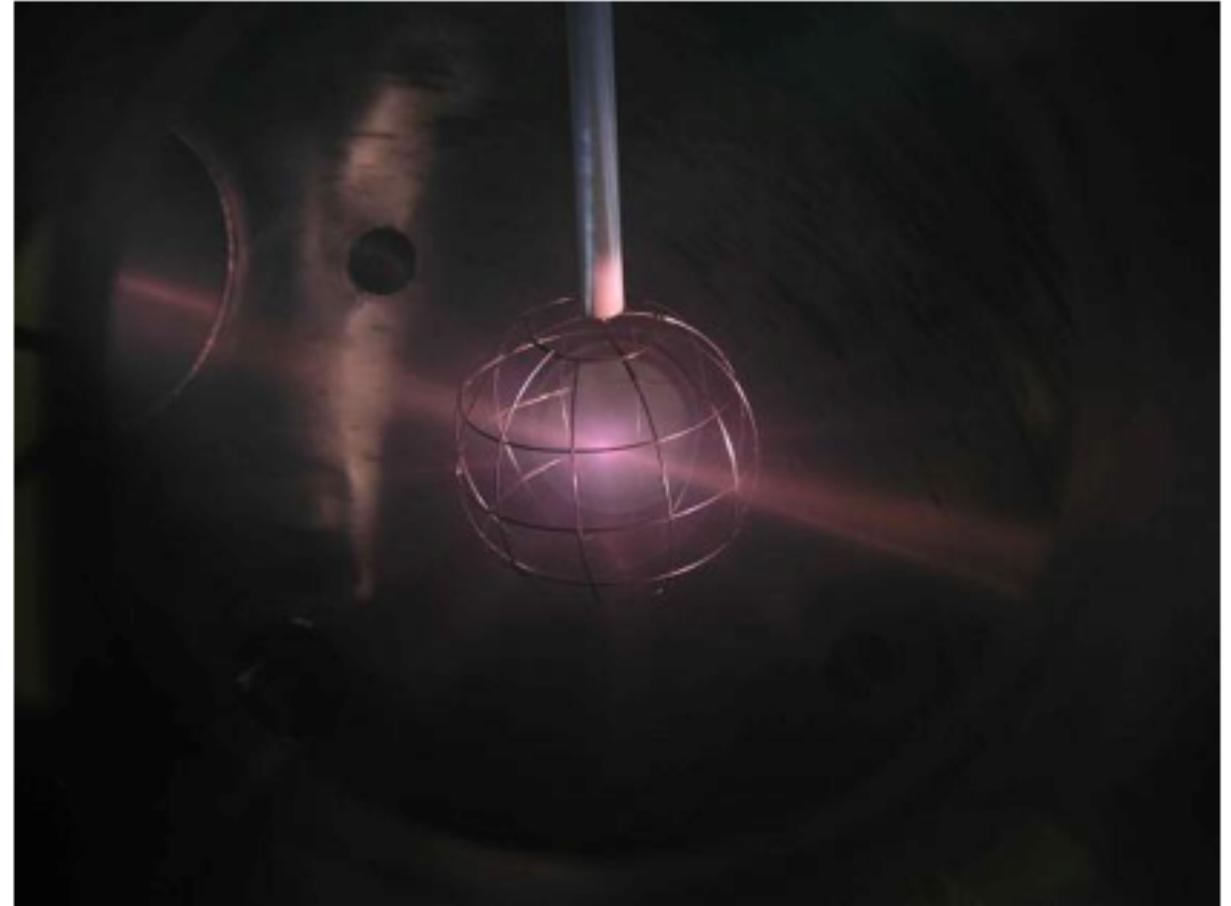
- Continued ^3He Fusion Research
- Technology to Access Large Sources of ^3He

Progress at Wisconsin Toward ^3He Fusion Reactors

D^3He fusion reactions in an IEC device demonstrated in 1999

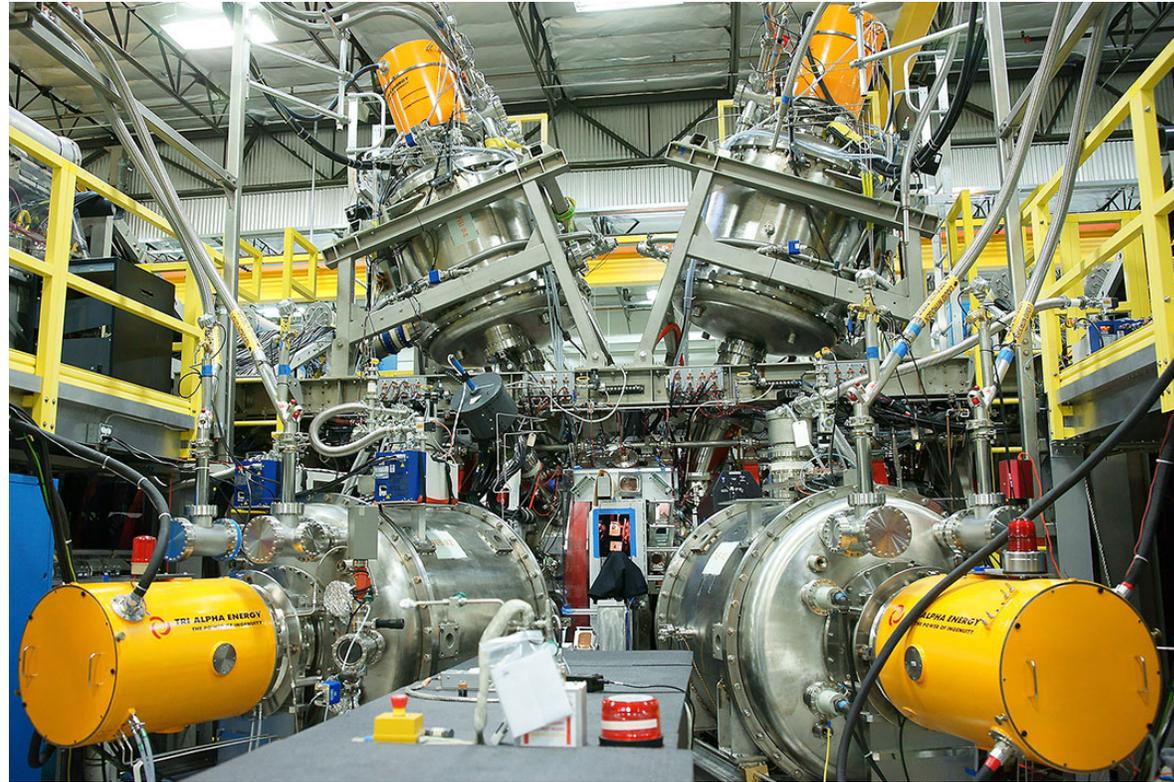


First demonstration of $^3\text{He}^3\text{He}$ reactions in a plasma in 2005

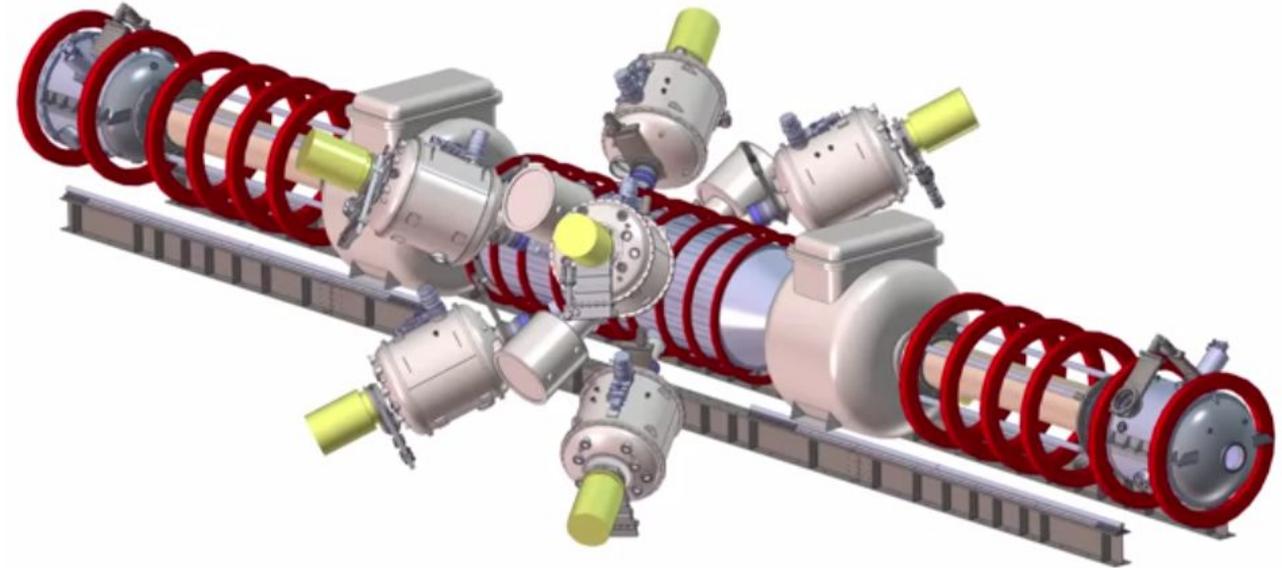


Commercial Progress in Advanced Fuel Fusion Reactors

Tri Alpha Energy, Lockheed Martin, Helion Energy, EMC2



Credit: Tri Alpha Energy, 2015



The ^3He on Earth is Insufficient to Support Commercial Fusion Power

There is <30 kg of available terrestrial ^3He (from Tritium Decay)

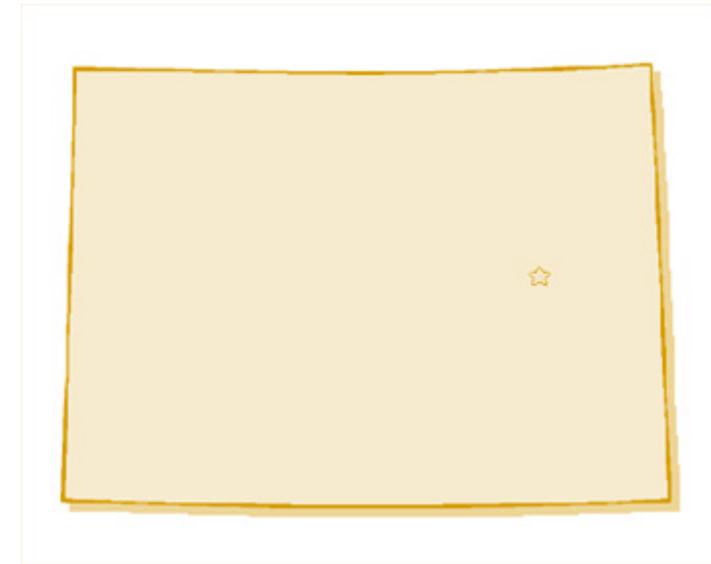
*0.6 GWyr of ^3He Fusion Energy

Enough for ^3He Fusion R&D

Only enough electricity for 5 weeks in Colorado

Colorado

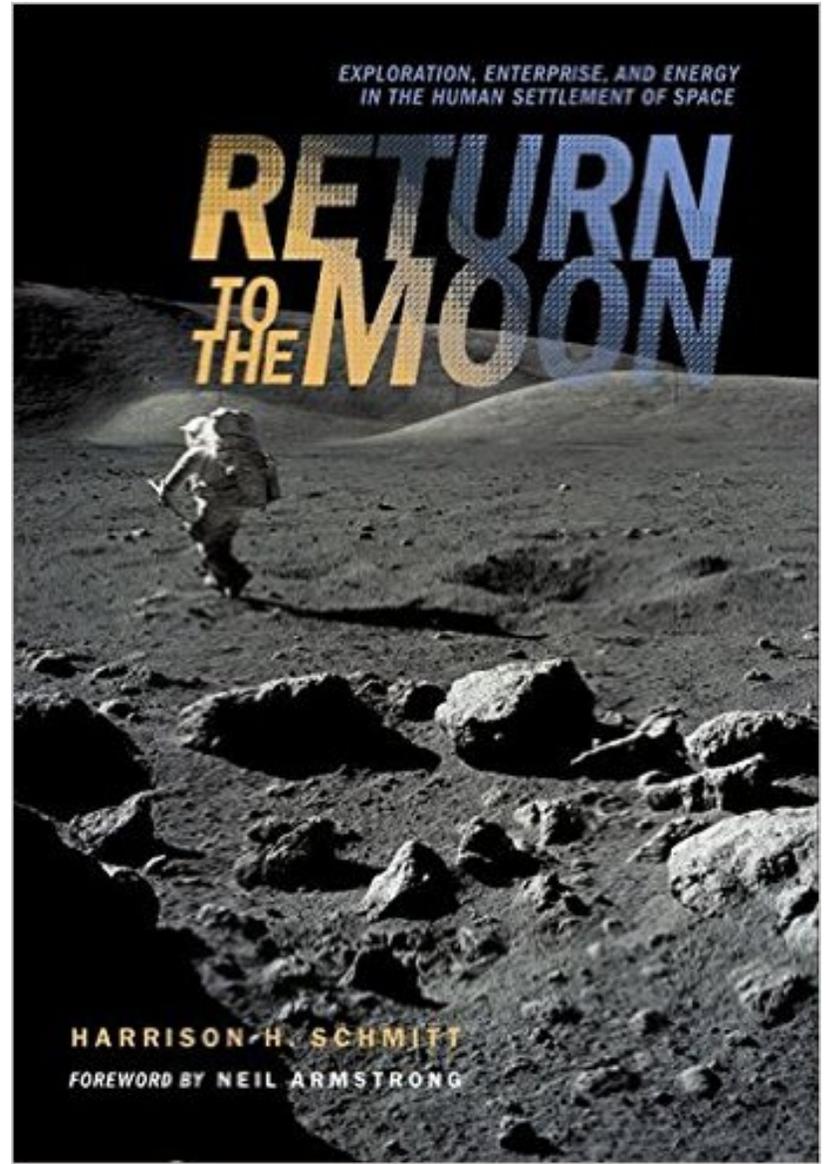
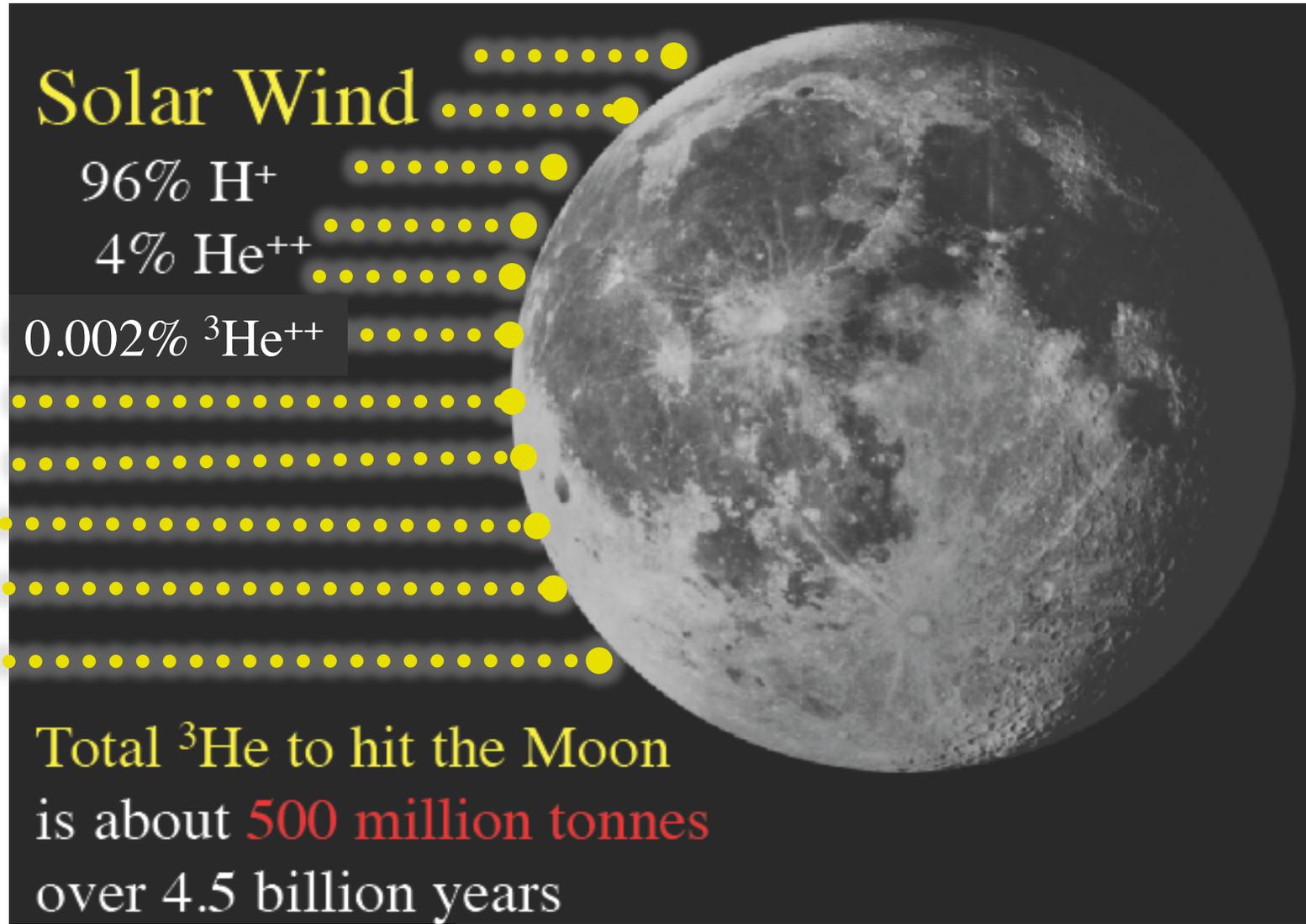
~6 GWyr electricity used in 2015



*20 MWyr/kg of fusion energy for ^3He



The Moon has retained over 1 million tonnes of ^3He



^3He on the Moon Could Become Extremely Valuable



At today's oil prices (\$38/barrel) the energy content in one tonne of ^3He would be worth \$3.7 billion

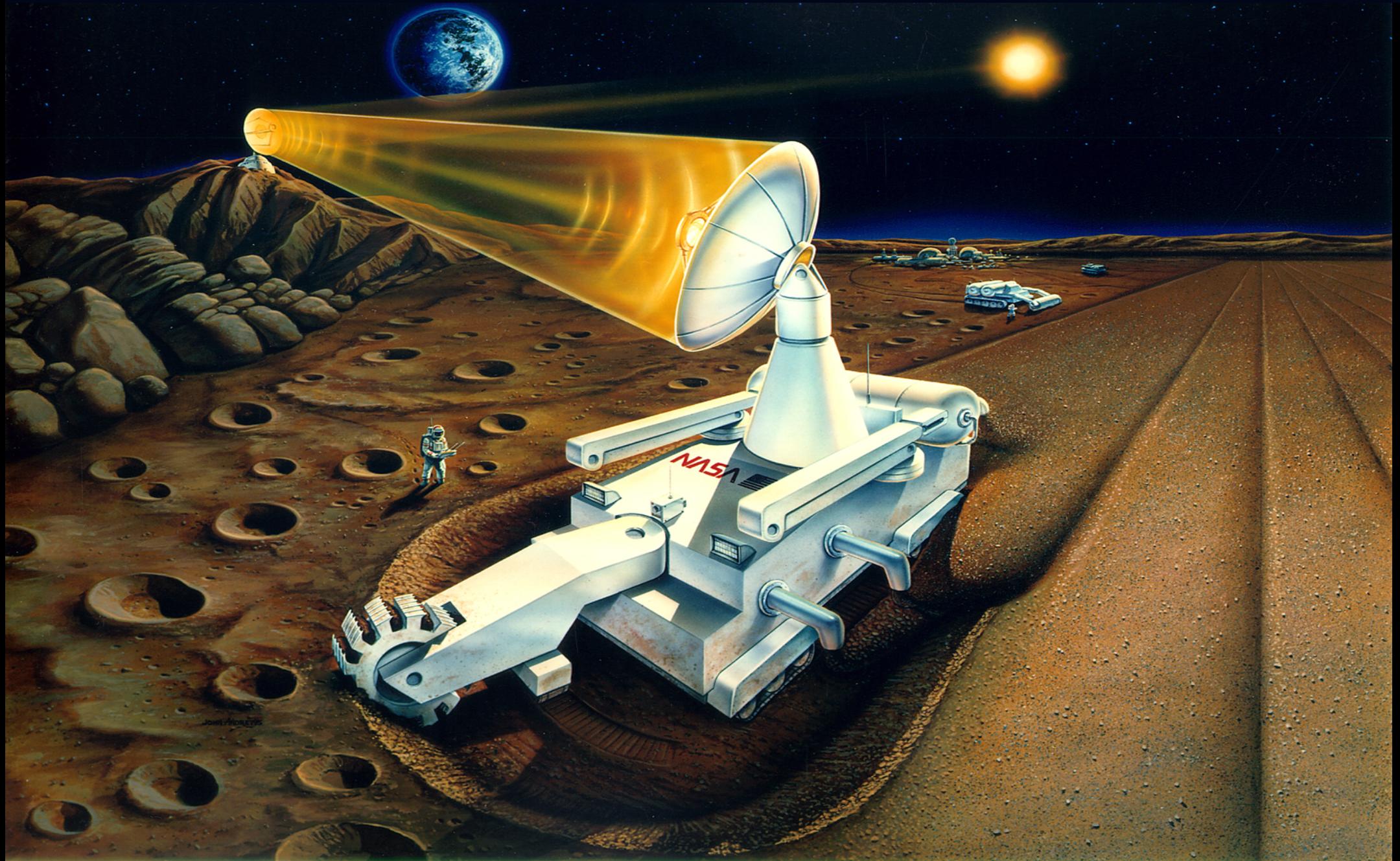
One of today's SpaceX Dragon capsules could return the volume of a ^3He "payload" worth around \$15 billion in energy.



Credit: SpaceX, 2012



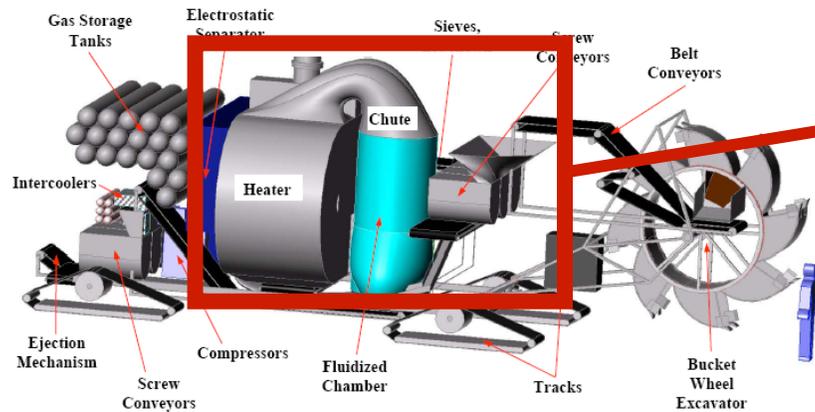
Three Iterations of Helium-3 Miners Designed at the UW FTI



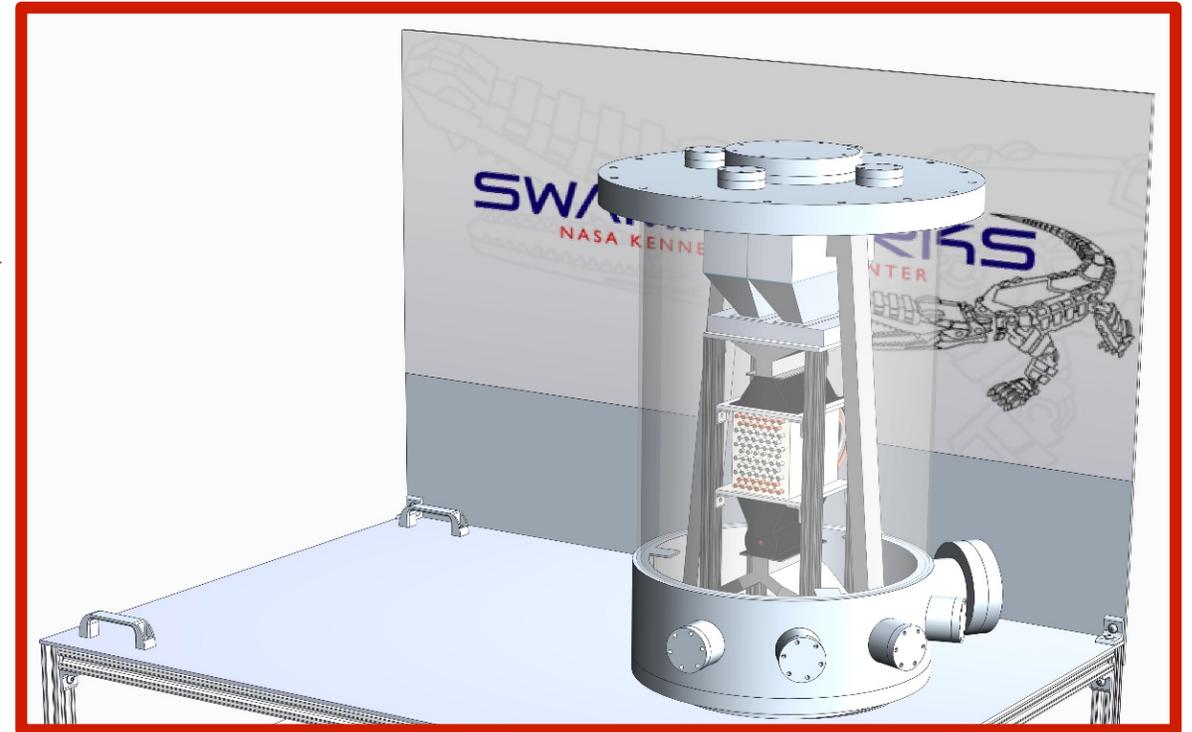
This Research Will Demonstrate ^3He Extraction

- First Demonstration of ^3He release inside of a heat pipe heat exchanger
- Laboratory scale (TRL 4)

Helium Extraction and Acquisition Test bed (HEAT)



Mark 3 Lunar Miner, Credit: M. Gajda



There are Two Main Aspects of this Work

- Implant He into Simulant
- 1kg Batches of Simulant
- Known Concentrations

Solar Wind Implanter (SWIM)

- Regolith Temperature to 700°C
- Heat Recovery Measurement
- He Release Measurement

**Helium Extraction and
Acquisition Testbed (HEAT)**



Solar Wind Implanter (SWIM) Objectives

Implantation Goals

- Implantation energy: ~ 1 keV/amu (solar wind)
- Aim for 20 ppb ^4He concentration
- ^4He diffuses out of regolith like ^3He (use ^4He for cost)

Implantation Characterization

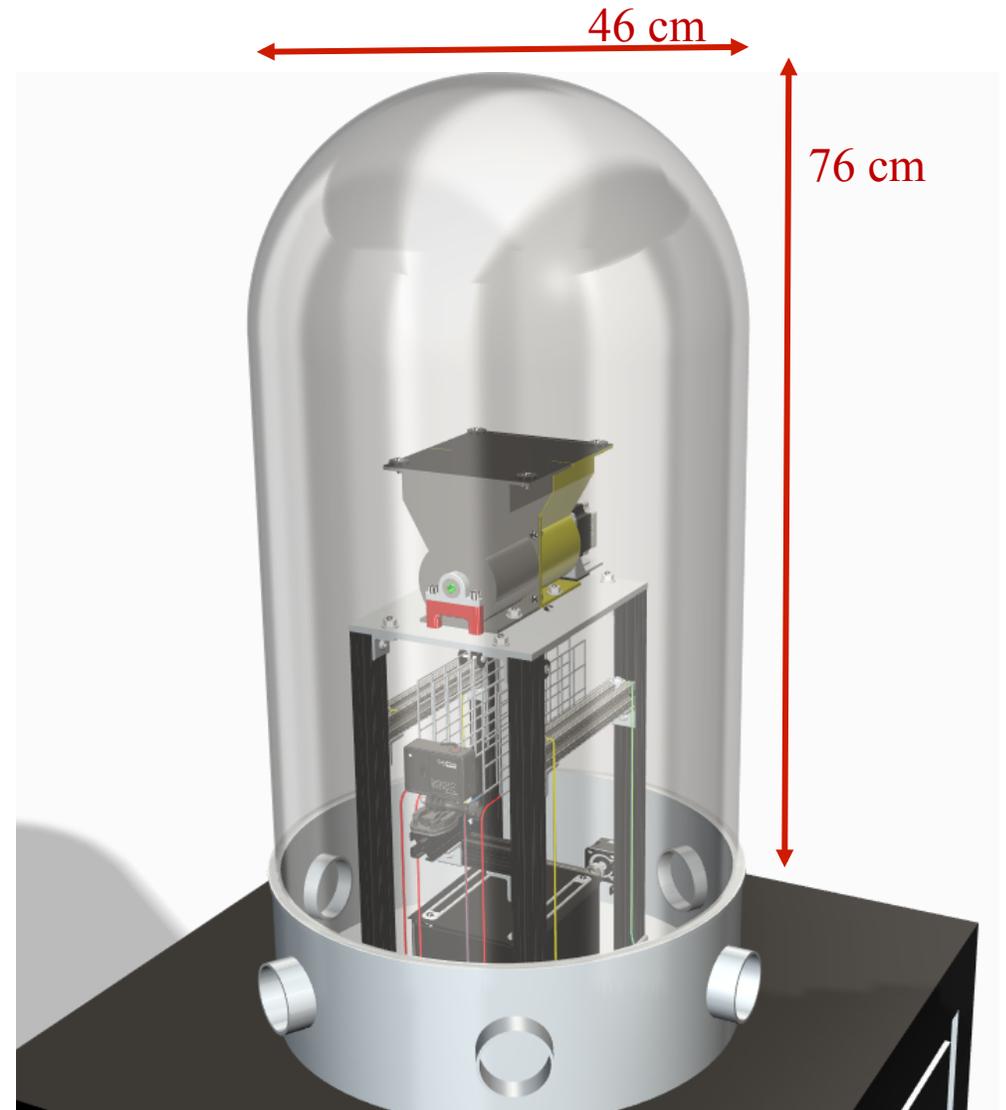
- Helium release vs. temperature
- Measure the released helium relative to any background ^4He concentration
- Energy spectrum of implanted ^4He

Past Lunar Regolith Implantation Studies

- Plasma Source Ion Implantation (PSII), helium was implanted into regolith simulant at energies between 5-50 keV/amu at the University of Wisconsin (Kuhlman, 1999 and 2012)
- More time consuming means of helium ion implantation into regolith minerals with ion beams have also been done (Futagami et al, 1990)
- An efficient and readily available solar wind implantation device for bulk batches of simulant did not exist to our best knowledge

SWIM Current Approach

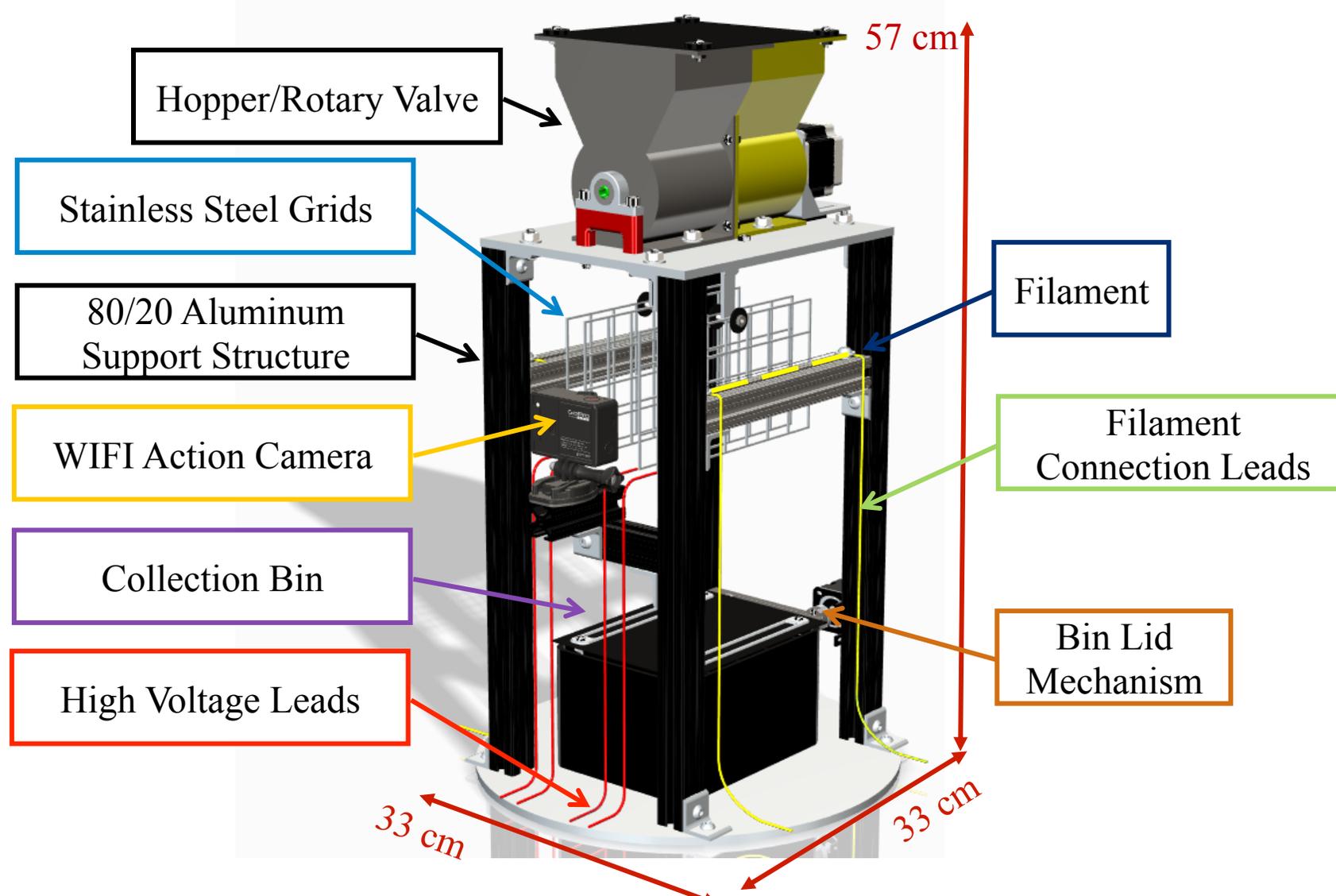
- Two sets of mesh cathode-anode pairs (implantation into both sides of the falling sheet)
- Use of tungsten filaments to ionize helium gas in vacuum chamber
- Rotary valve (stepper motor driven) to discharge simulant in as thin of falling “sheet” as possible
- Camera to record simulant flow through plasma of inside chamber



Current SWIM CAD Model



SWIM Components



SWIM: Implantation Analysis

Mass of helium to be implanted

$$M_{\text{He}} = \left(\frac{c}{1-c} \right) M_{\text{sim}}$$

Helium ion flux

$$\Gamma_{\text{He}} = \frac{N_A c \dot{M}_{\text{sim}}}{2W_{\text{He}}(1-c)(1-F_o)L_{\text{grid}}h_{\text{grid}}}$$

Helium ion current density

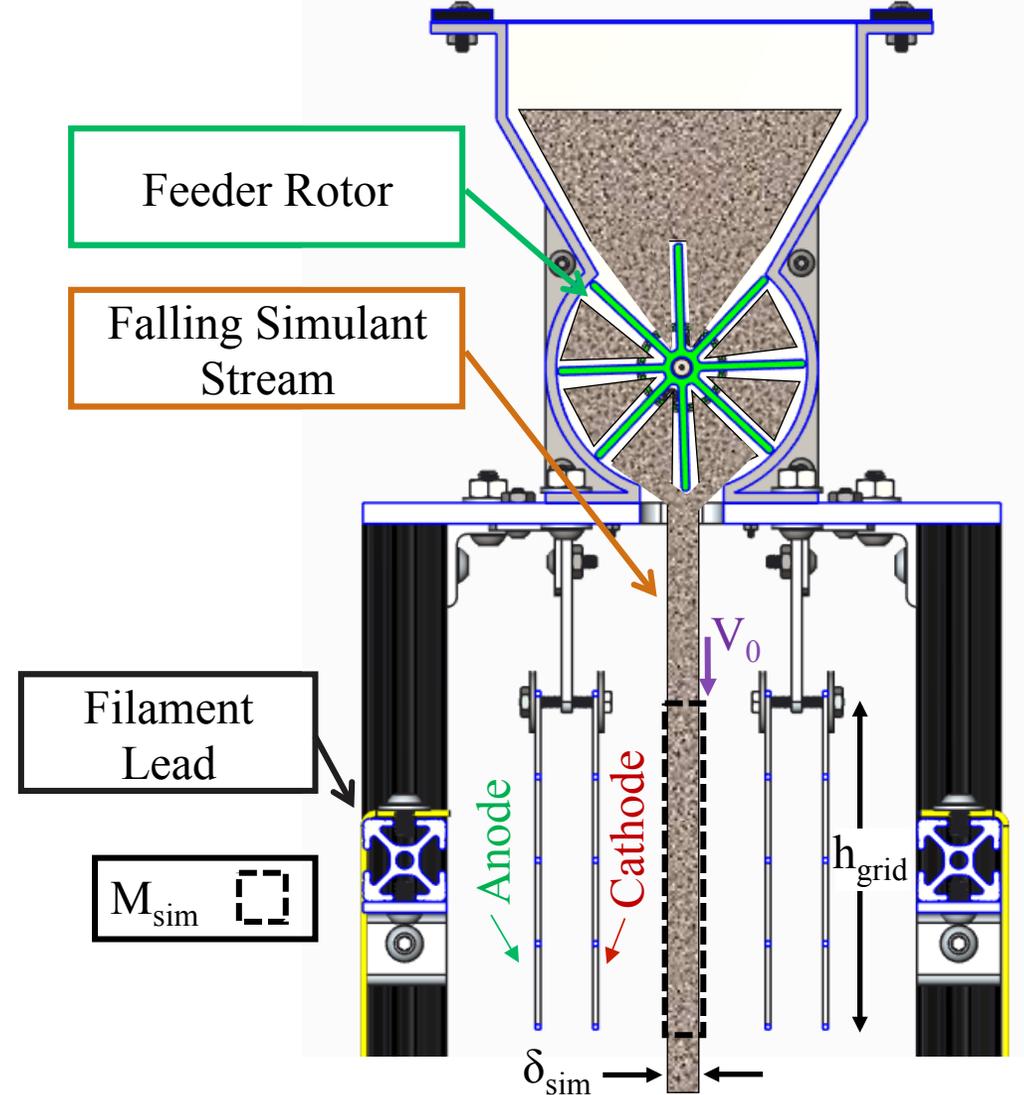
$$J_{\text{imp}} = e\Gamma_{\text{He}}$$

Measured current

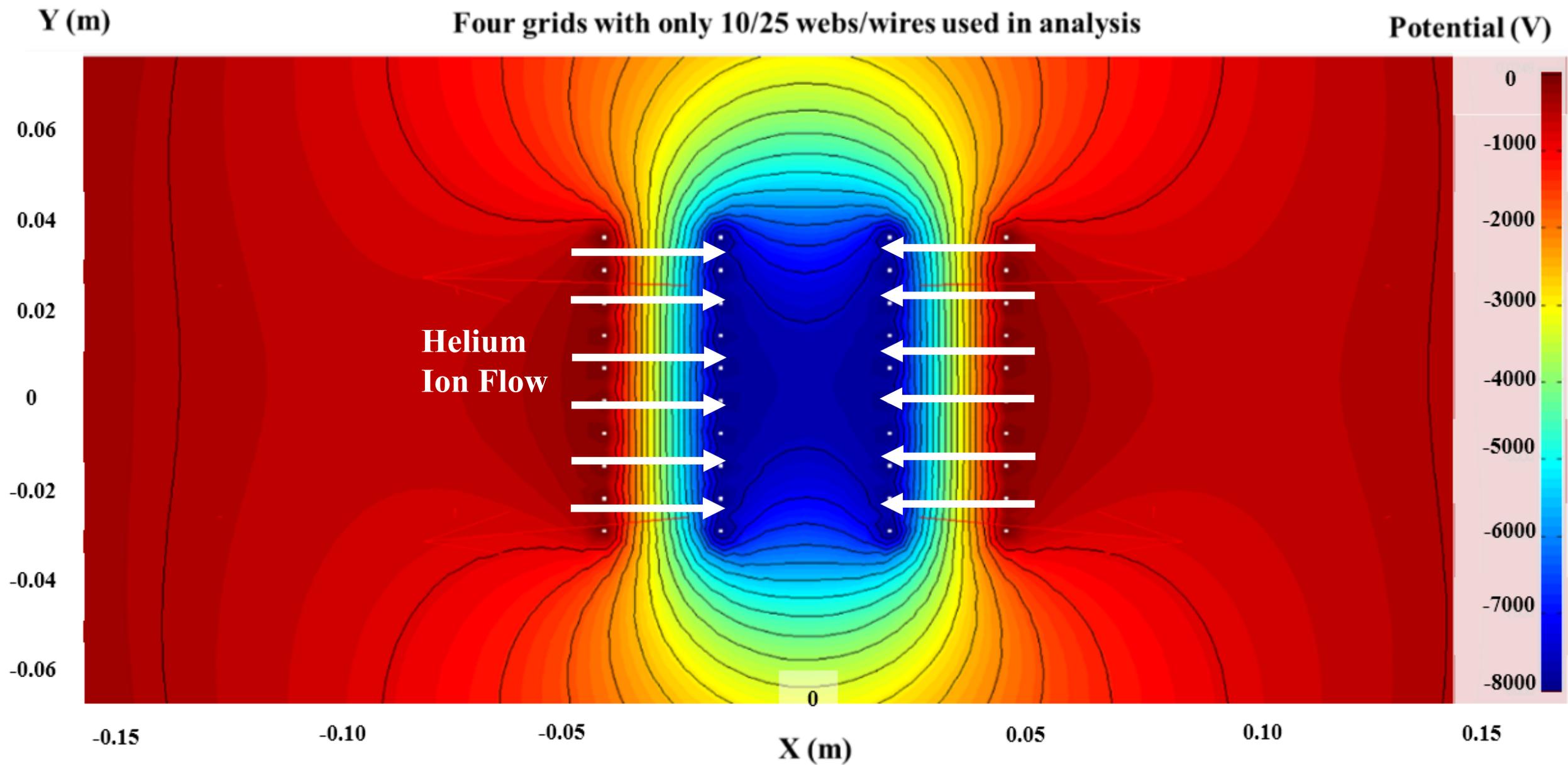
$$I_{\text{meas}} = I_{\text{ion}}(1+\gamma)$$

Simulant stream thickness

$$\delta_{\text{sim}} = \frac{\dot{M}_{\text{sim}} \left(-v_0 + \sqrt{v_0^2 + 2gh_{\text{grid}}} \right)}{g\rho L_{\text{sim}} h_{\text{grid}}}$$

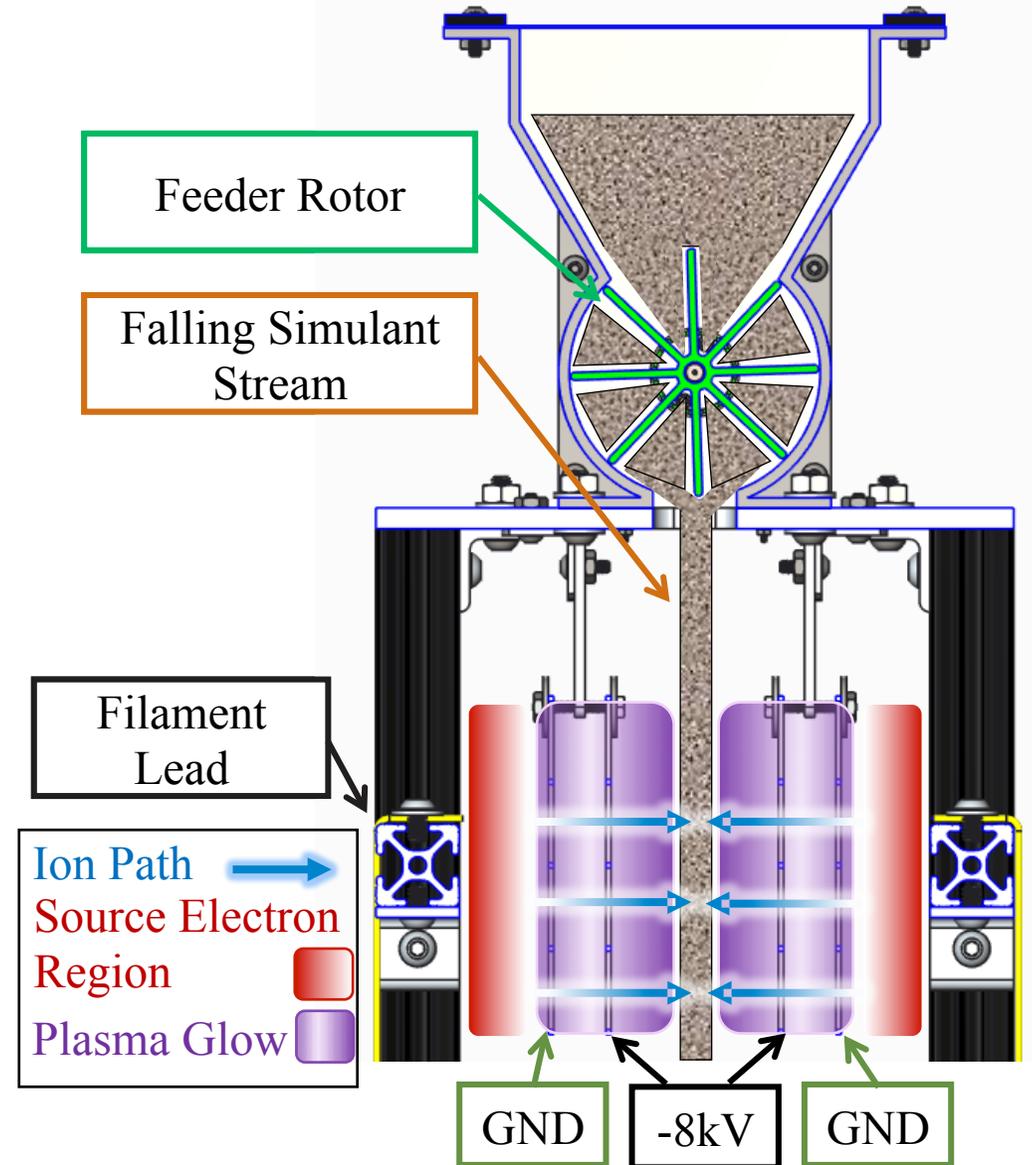


SWIM: Acceleration Grid Assembly



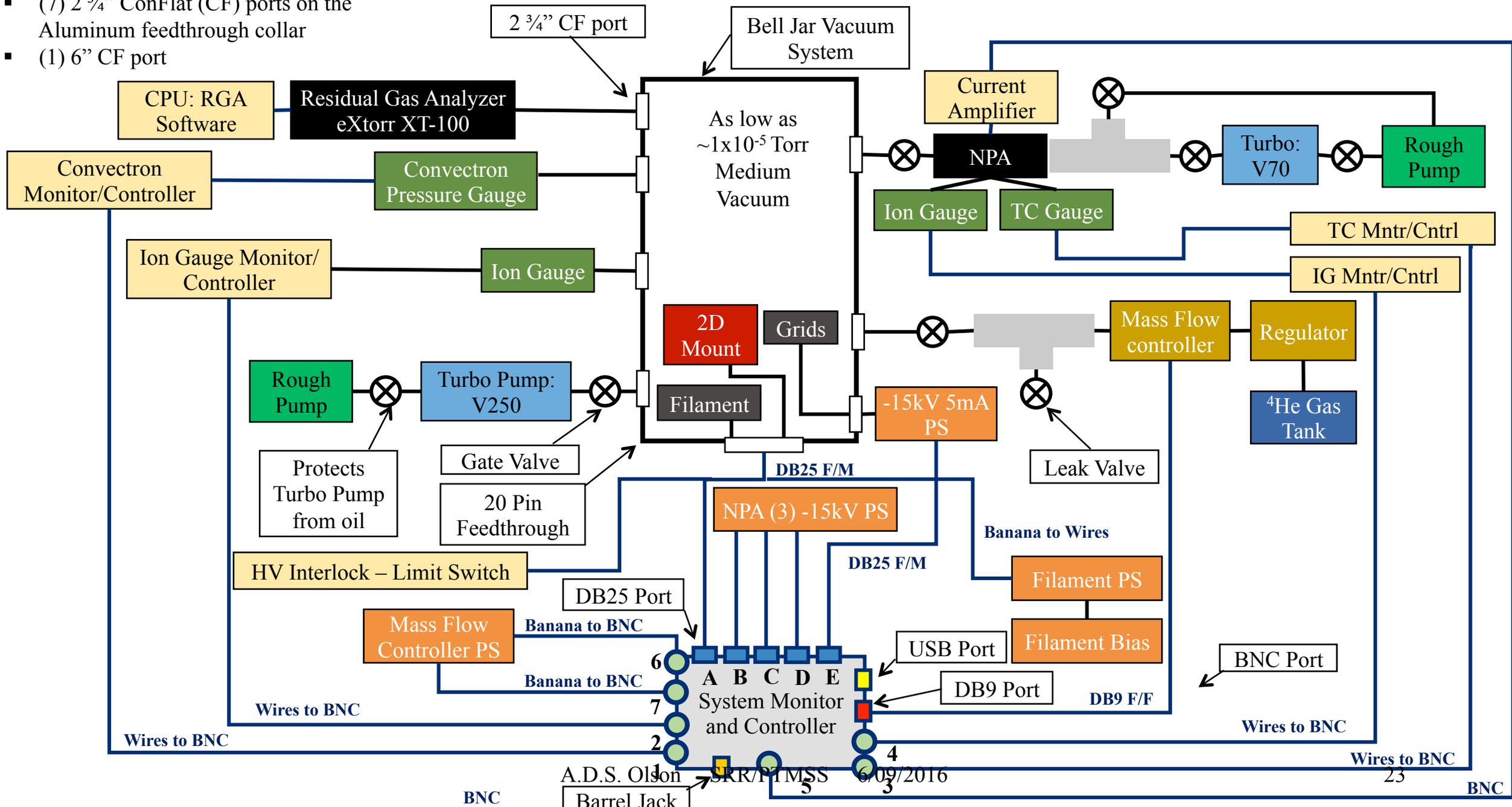
SWIM: Device Operation

- Fill hopper with up to ~2 kg of JSC-1A
 - (<100 micron)
- Pressure: $\sim 1 \times 10^{-4}$ Torr helium-4
- Grid voltage and current set
- Drive rotor to empty hopper
- Action camera records simulant flow and plasma glow
- Grid and filament voltages and currents recorded

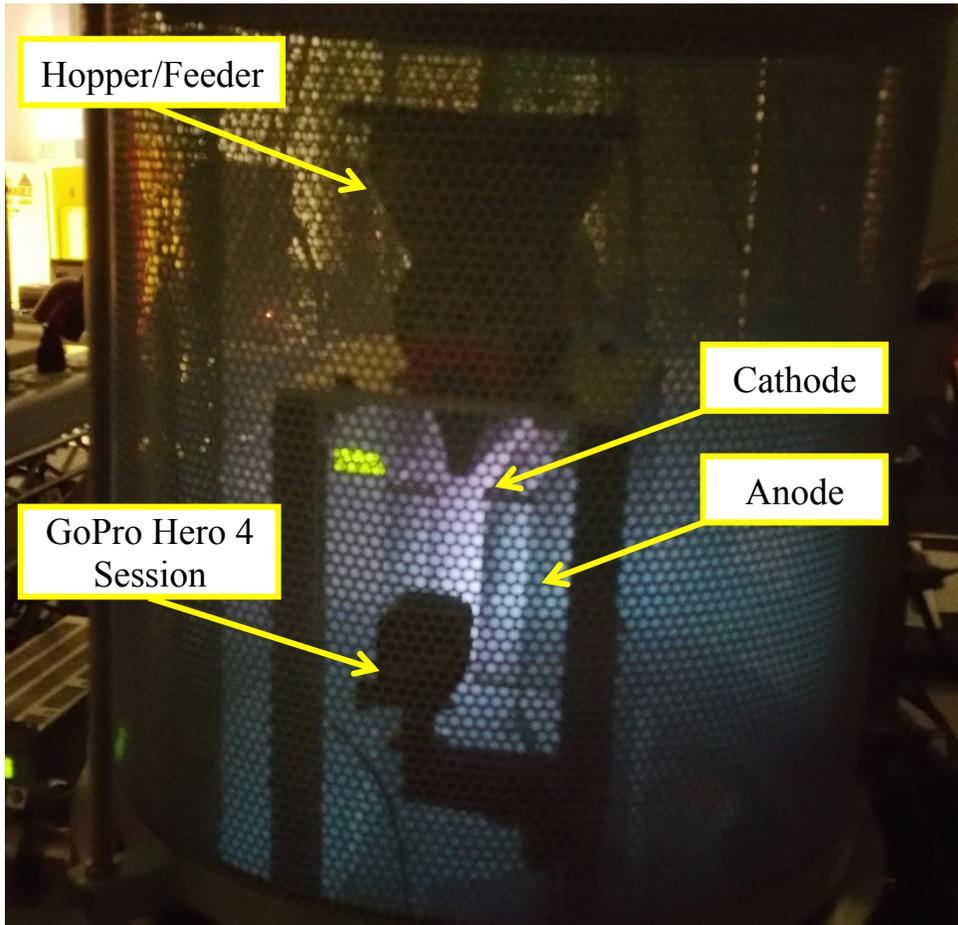


SWIM NPA Layout A.D.S. Olson (FTI, Univ. of Wisconsin) 04/28/2016

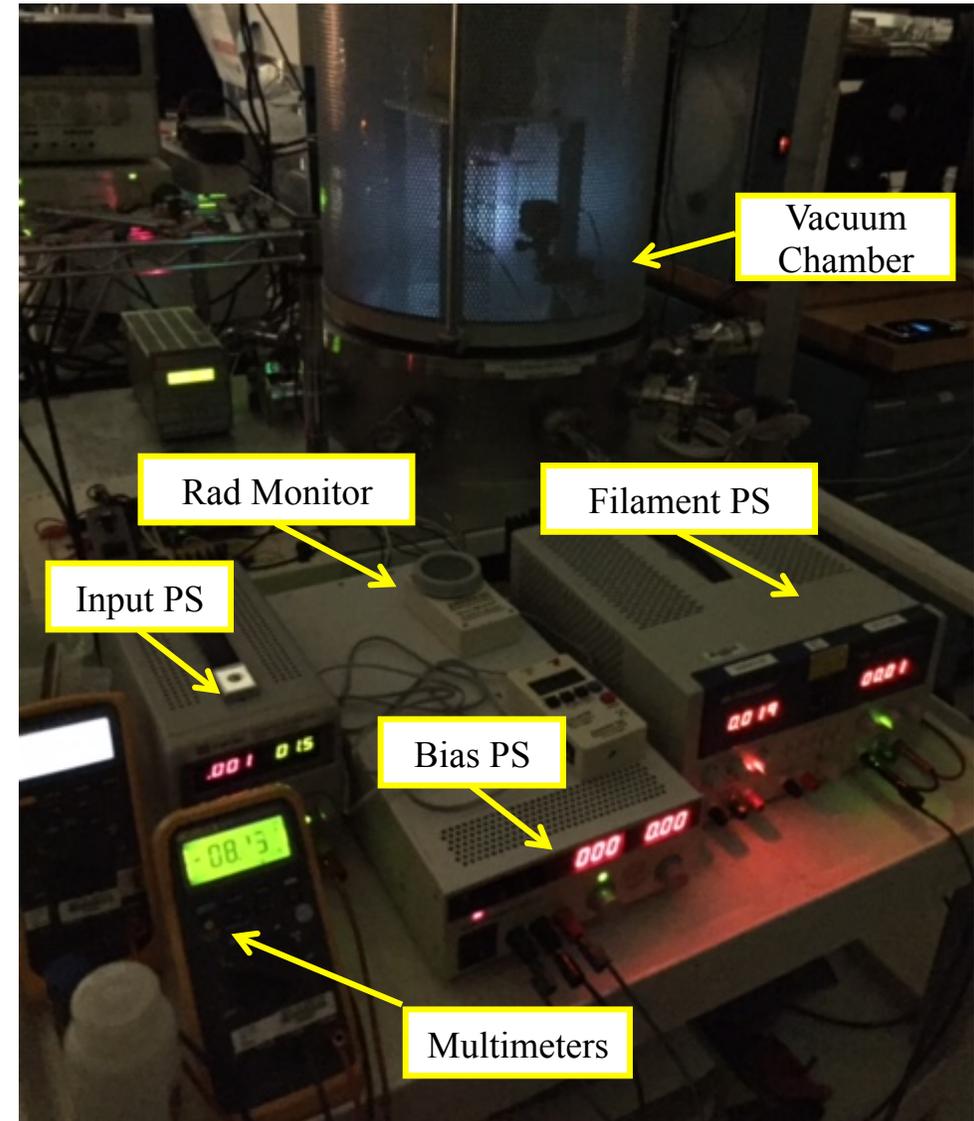
- 18" OD, 22" tall glass cylinder
- (7) 2 3/4" ConFlat (CF) ports on the Aluminum feedthrough collar
- (1) 6" CF port



SWIM: Setup at NASA KSC ESPL



- HV amplifier ($\pm 10\text{kV}$, 10mA)
- HV input power supply
- Filament bias power supply (600V , 1.7A)
- Filament power supply (30V , 4A)
- Multimeters to measure output voltage and current
- AdAware radiation monitor
- Varian 301 turbo pump, gate valve, and controller
- Scroll pump
- Extorr 300 RGA
- 18"x30" bell jar and 8-port feed through collar
- Helium K bottle
- Baratron gauges
- Ion gauge



Pressure = 0.08mTorr ^4He
Grid Voltage = 8kV
Grid Current = 4.96mA
Filament Voltage = 19.9V
Filament Bias = -300V

72.3 g of <100 micron
JSC-1A simulant
implanted

2.5cm

Funnel

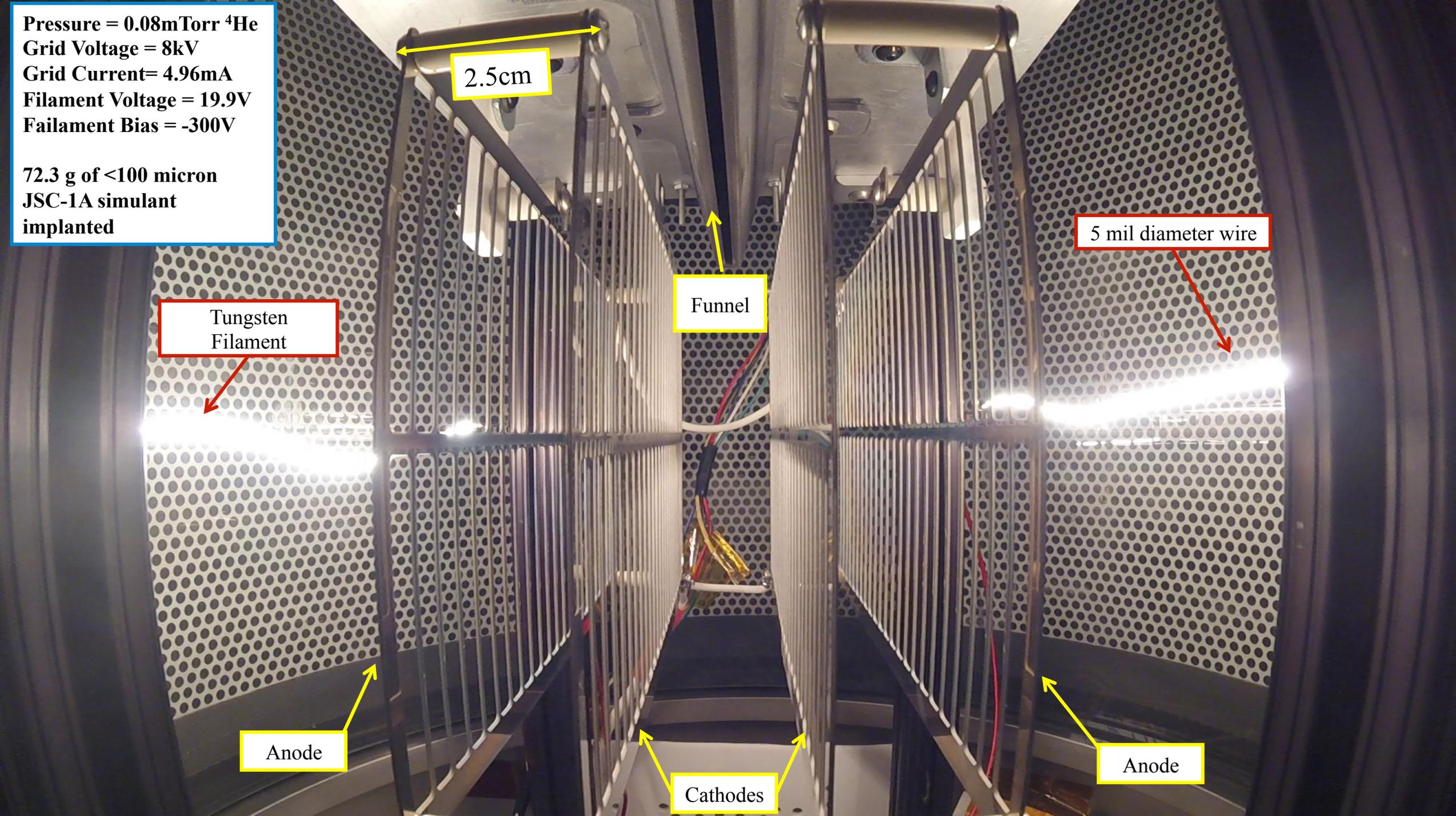
5 mil diameter wire

Tungsten
Filament

Anode

Cathodes

Anode



Pressure = 0.08mTorr ⁴He
Grid Voltage = 8kV
Grid Current = 4.96mA
Filament Voltage = 19.9V
Filament Bias = -300V

72.3 g of <100 micron
JSC-1A simulant
implanted

2.5cm

Tungsten
Filament

Simulant
Flow

5 mil diameter wire

Anode

Cathodes

Anode



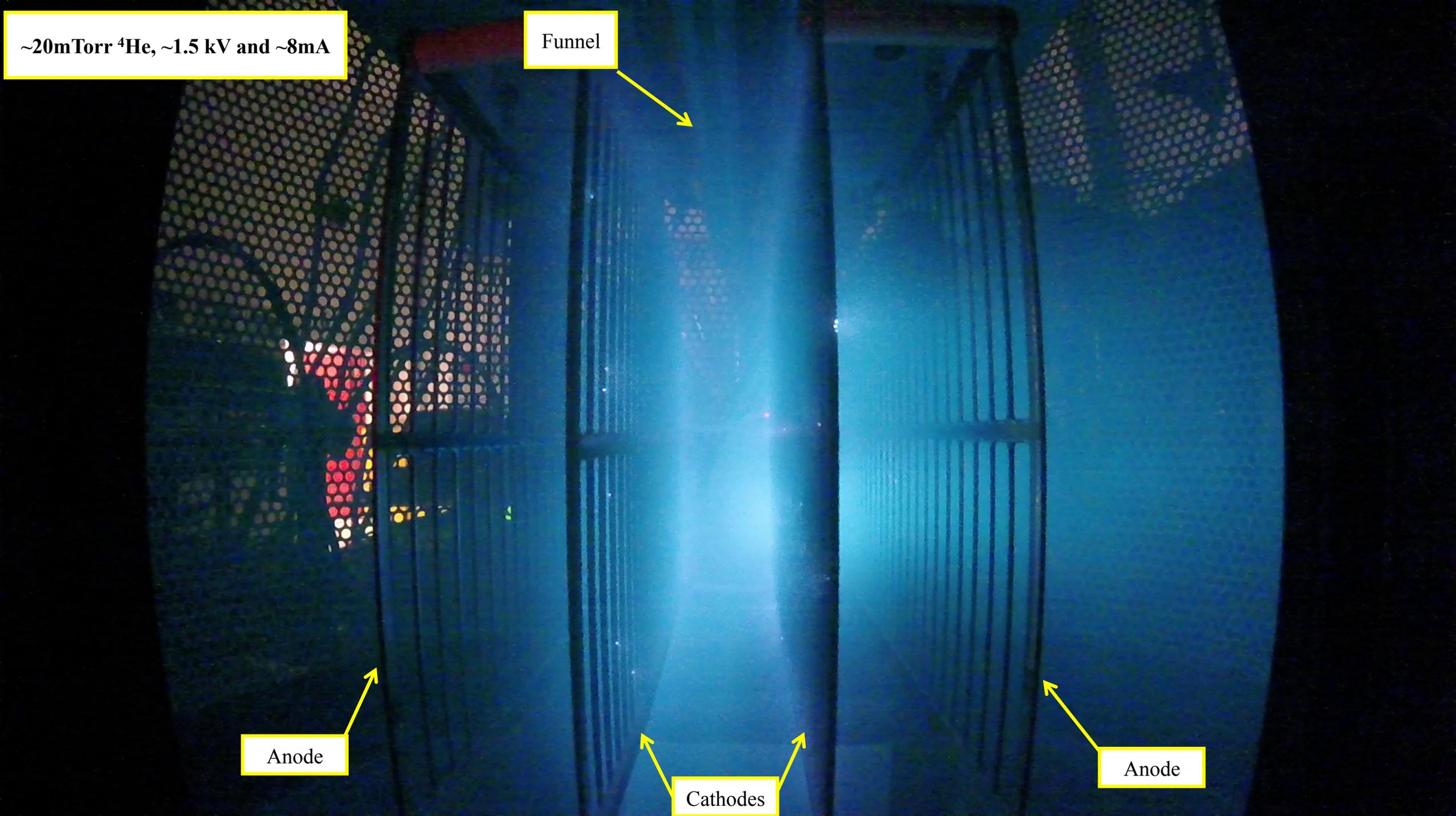
~20mTorr ^4He , ~1.5 kV and ~8mA

Funnel

Anode

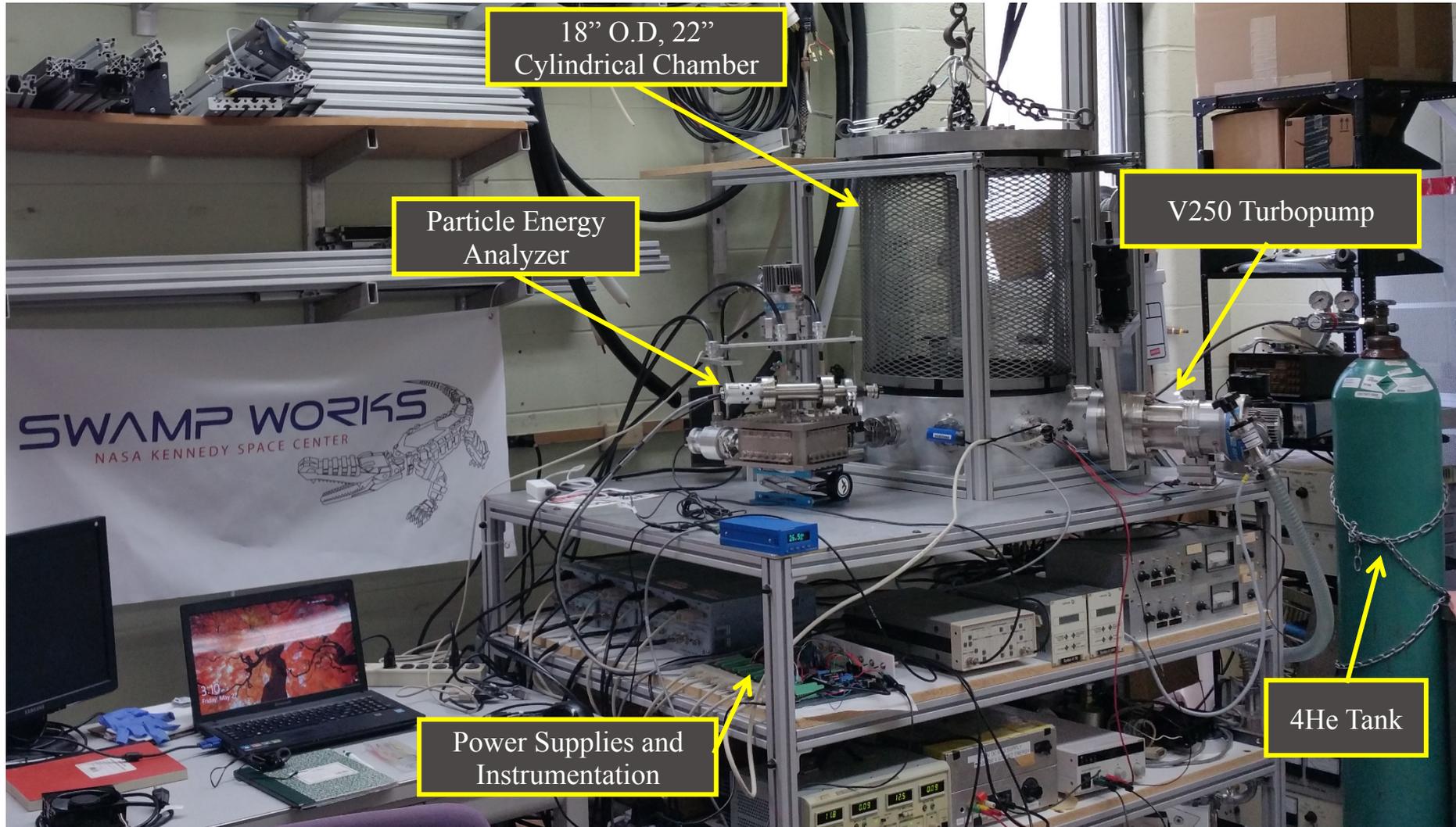
Cathodes

Anode



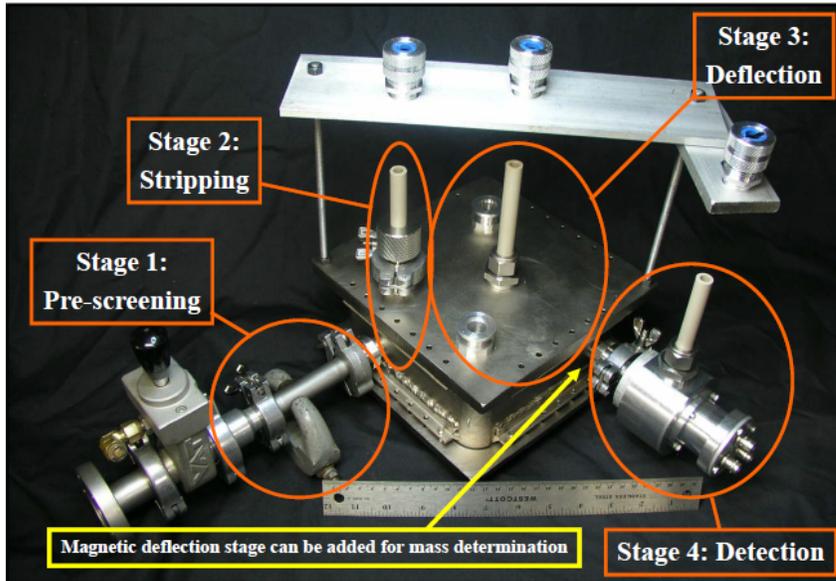


SWIM Setup Replicated in FTI Lab for Continued Testing

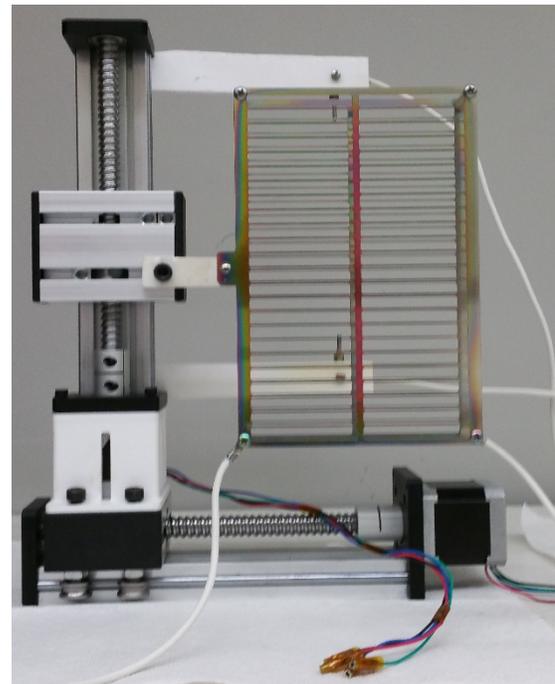


Helium Ion Energy Spectrum Testing

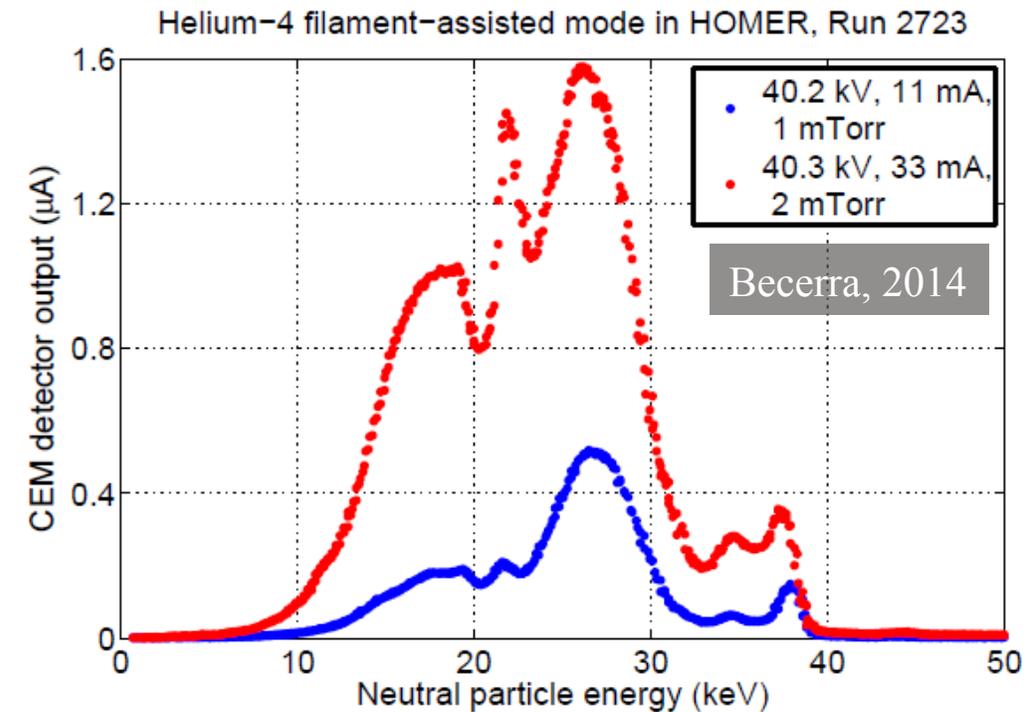
- Attach particle energy analyzer to SWIM chamber
- Mount SWIM grid assembly onto a moveable (2 DOF) platform
- Measure $^4\text{He}^+$ energy spectrum: vary pressure, grid voltage, filament settings



Energy analyzer detached from vacuum chamber

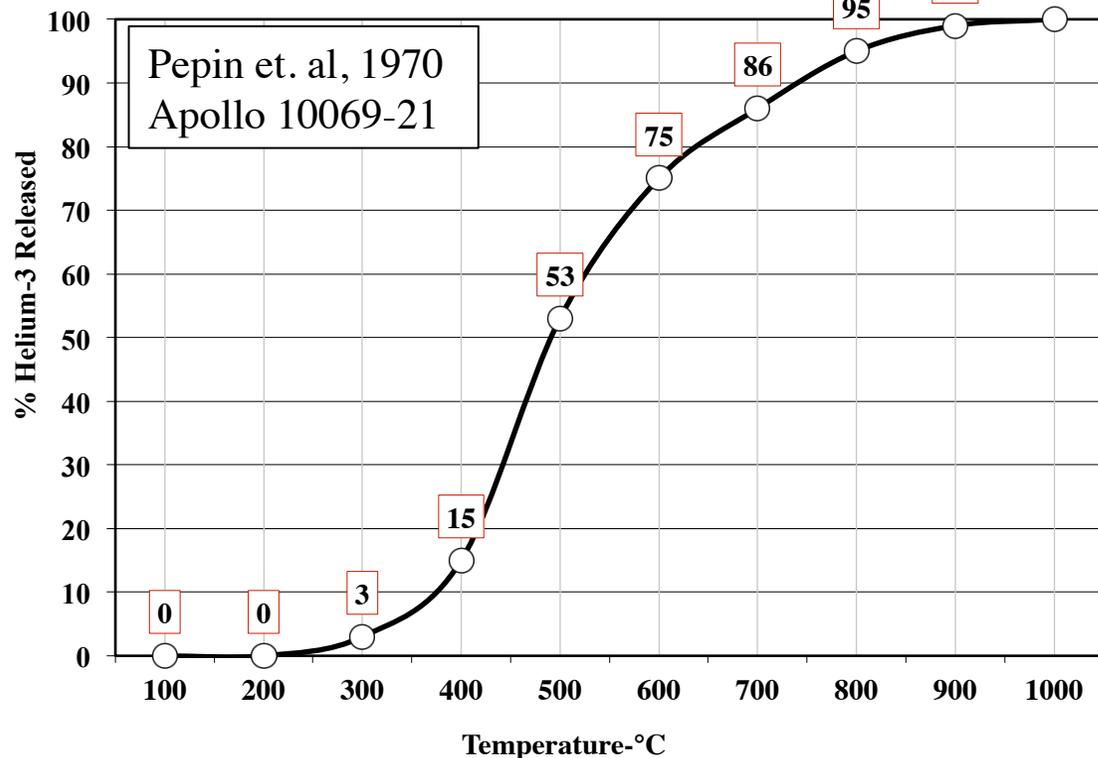


SWIM grids on 2D platform

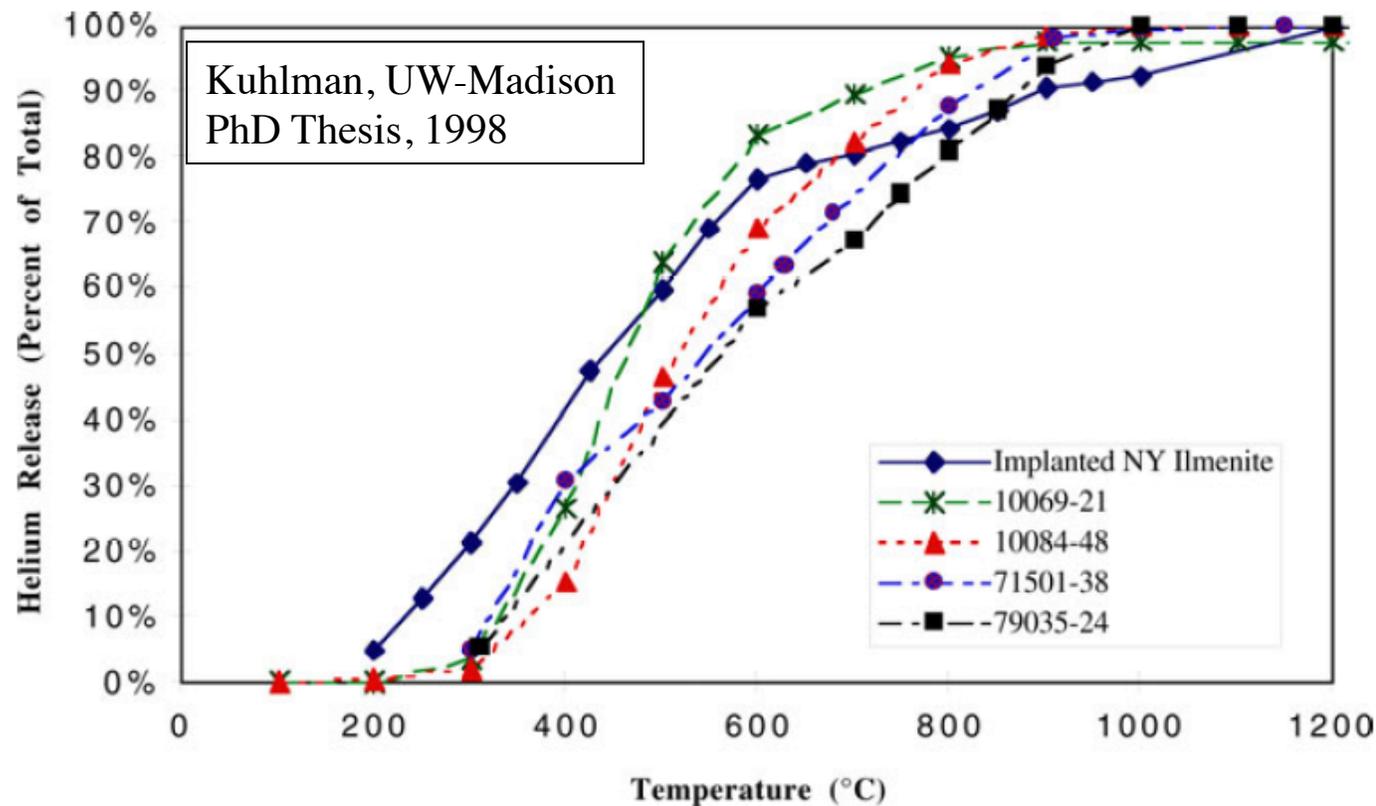


Helium Release vs. Temp: Dr. Pepin at U. Minnesota

Lunar Regolith



Analog Regolith



Application to General Space Resources Field

SWIM

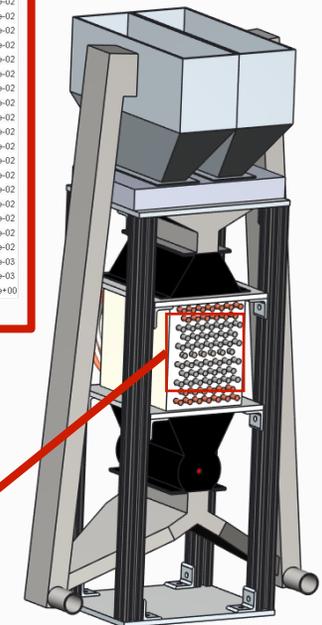
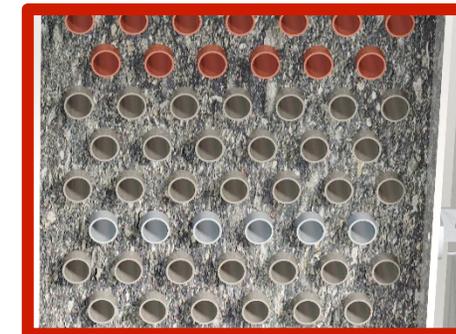
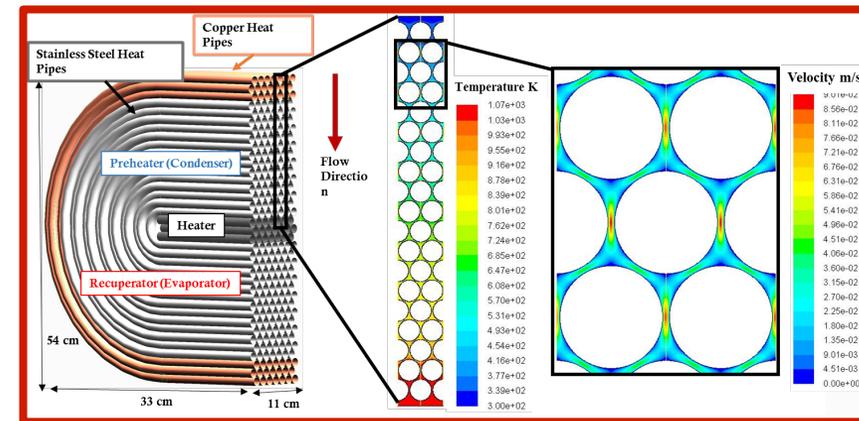
- Implantation of other volatiles
- Simulation of space weathering effects

HEAT

- Heat recuperation with heat pipes
- Granular flow in heat exchangers
- Volatile gas release measurement with residual gas analyzers



JSC-1A (Orbitec)

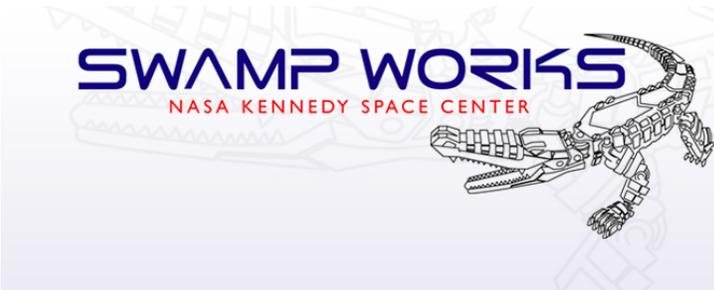


Conclusion: Progress and Future Work

- Conclusions
 - SWIM designed and can produce implanted simulant
 - More work to go to better characterize the implantation
- Up Next:
 - Complete ion energy spectrum tests
 - Sample testing at Univ. of Minnesota
 - More batches created for use in the HEAT system
 - Possible implantation of other gases for future work

This Research is Supported by a NASA Fellowship

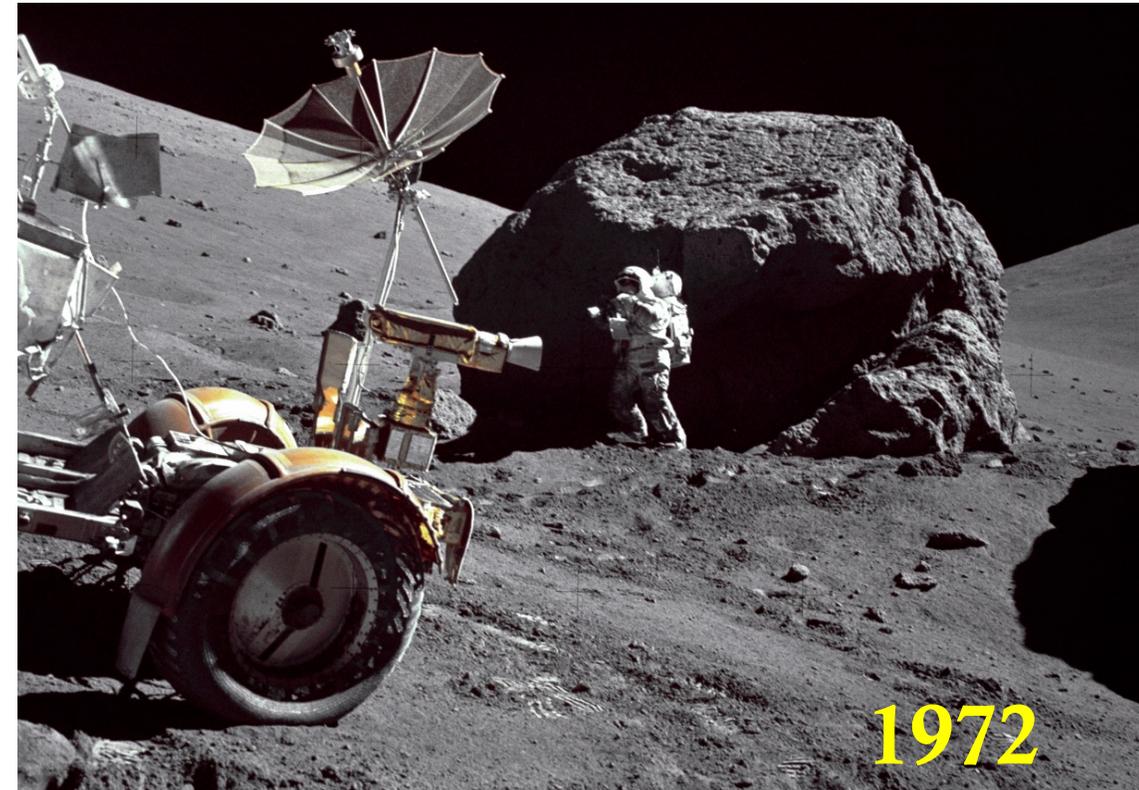
- Space Technology Research Fellowship (NSTRF)
- Class of 2014
- Collaboration with Kennedy Space Center



A screenshot of the NASA website. The top navigation bar includes links for NEWS, MISSIONS, MULTIMEDIA, CONNECT, and ABOUT NASA. Below this is a search bar and a secondary navigation bar with links for "For Public", "For Educators", "For Students", and "For Media". The main content area displays a news article titled "Lunar Volatiles Extraction Technology for Future Fusion Power and Multi-Outpost Scale Human Space Exploration" dated December 8, 2014, by Aaron Olson from the University of Wisconsin, Madison. The article text describes a prototype lunar volatiles extraction system. To the right of the article is a portrait of Aaron Olson. On the left side of the page, there is a sidebar menu for the "Space Technology Mission Directorate" with various sub-links.

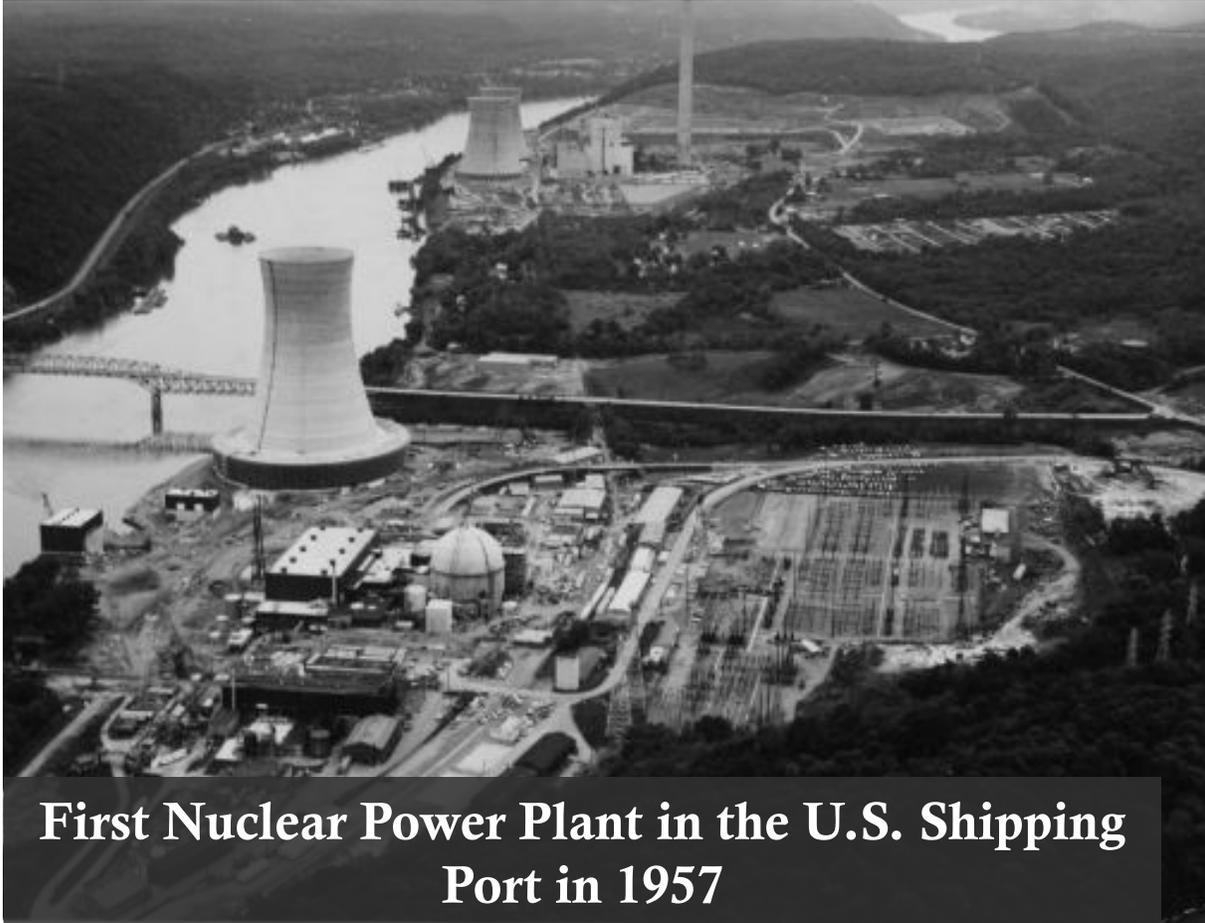
Mining ^3He Could Happen Sooner than you Think

“Space travel is utter bilge.” –Dr. Richard Wooley, Astronomer Royal,
space advisor to the British government, 1956

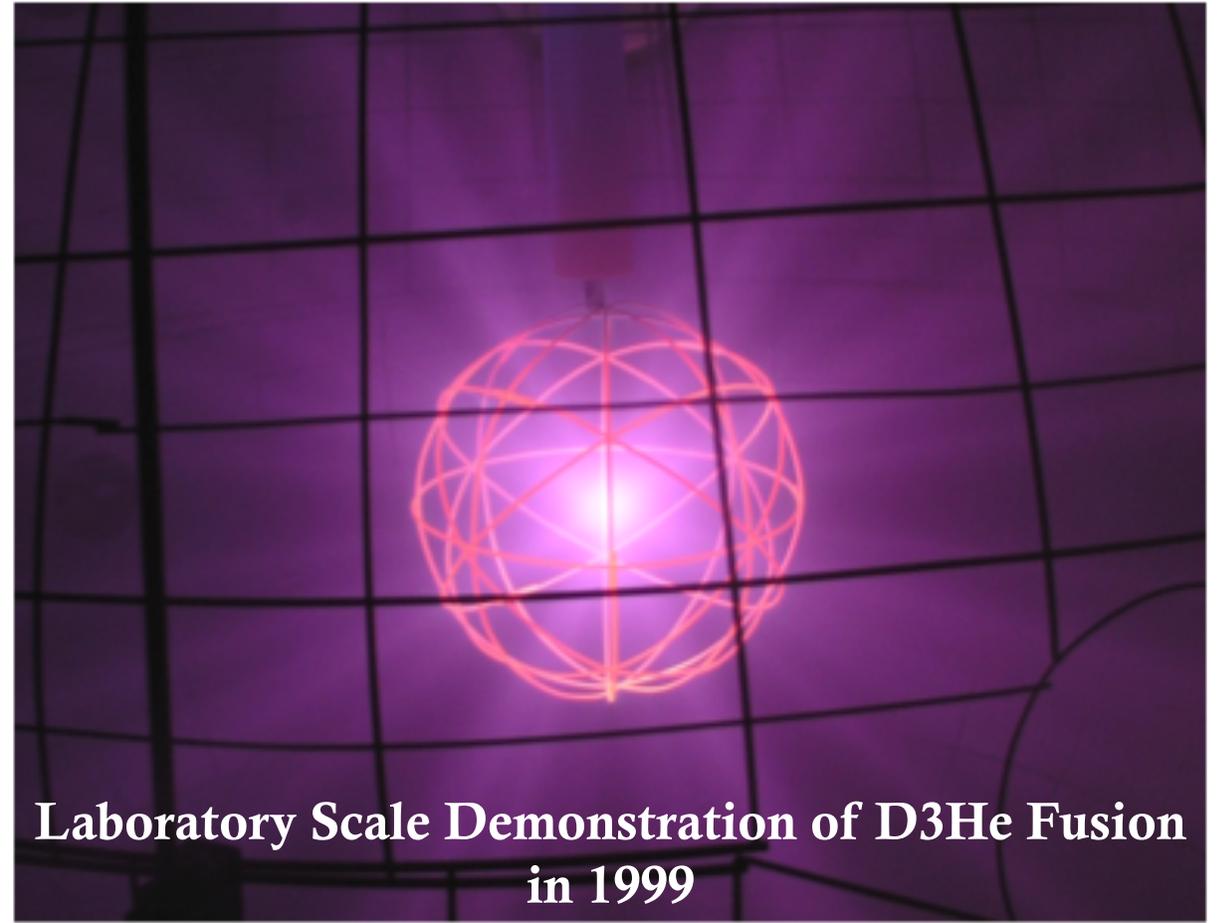


Fusion Could Happen Sooner than you Think

“Anyone who looks for a source of power in the transformation of the [nucleus of the] atom is talking moonshine.” –Ernest Rutherford, 1933



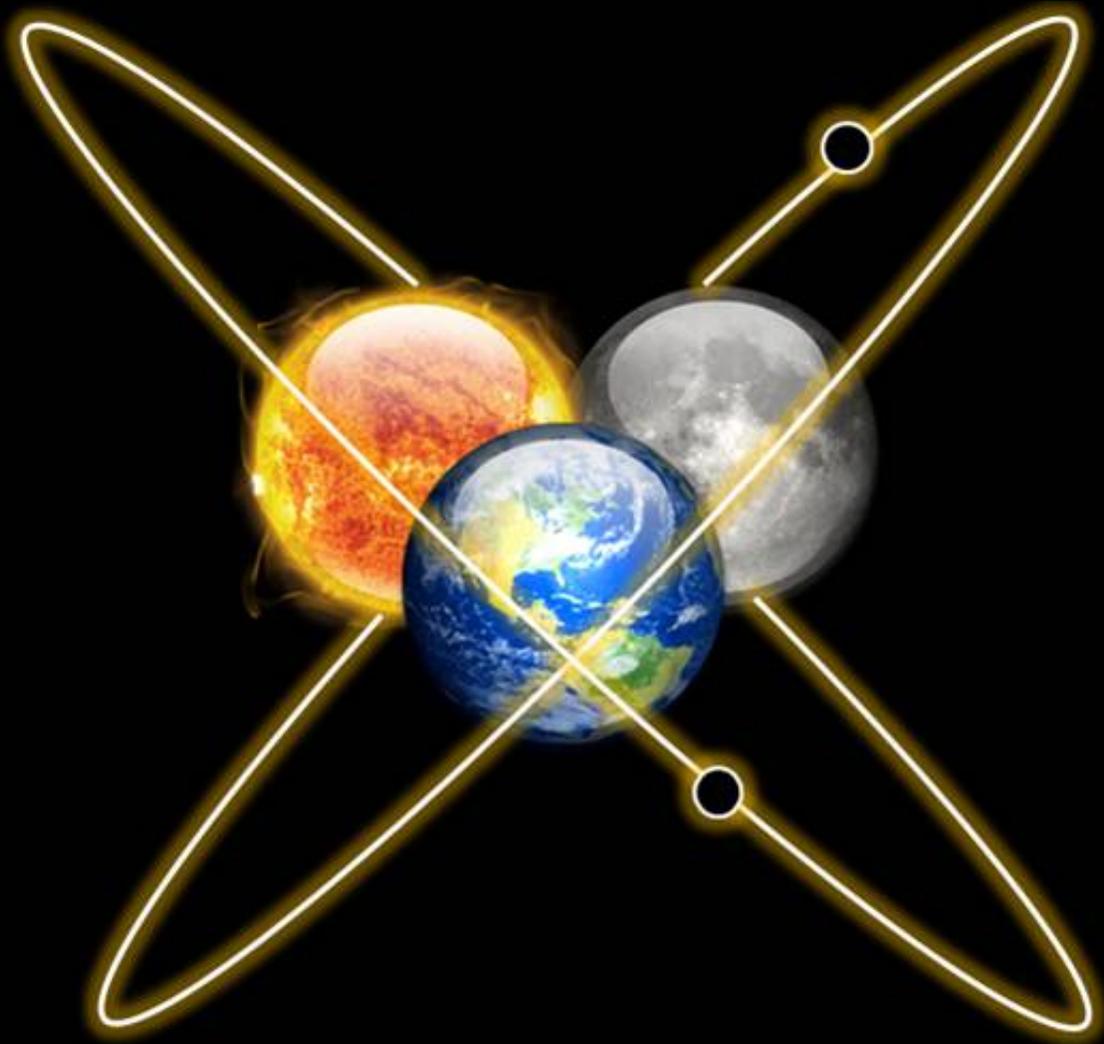
First Nuclear Power Plant in the U.S. Shipping Port in 1957



Laboratory Scale Demonstration of D³He Fusion in 1999



Questions?



Acknowledgments

- NASA Space Technology Research Fellowship Program
- NASA Kennedy Space Center Swamp Works Labs
- Wisconsin Space Grant Consortium
- Alex Strange, Tashi Atruksang

LUNAR HELIUM-3

THE EARTH'S ENERGY FUTURE

Key References

■ Fusion Related:

- Wittenberg, L. J., Santarius, J. F., and Kulcinski, G. L. (1986). “Lunar Source of ^3He For Commercial Fusion Power.” *Fusion Technology*, 10(September), 167–178.
- Wittenberg, L. J. (1989). “Terrestrial Sources of Helium-3 Fusion Fuel- A Trip to the Center of the Earth.” *Fusion Technology*, 15(1108).
- Santarius, J. F. (2013). “Overview of UW FTI Inertial-Electrostatic Confinement (IEC) and Other Research.” *Plasma Physics Seminar, University of Wisconsin, 18 February 2013, 2*.
- Kulcinski, G. L. (2012). “Fusion Fuel Resources.” *UW-Madison, NEEP-536 Course Lecture 6, Sept. 20, 2012, 20*.

■ Lunar ^3He Extraction Related:

- Sviatoslavsky, I. N., and Jacobs, M. K. (1988). “Mobile Helium-3 Mining and Extraction System and Its Benefits Toward Lunar Base Self-Sufficiency
- Sviatoslavsky, I. N. (1993). “The Challenge of Mining He-3 on the Lunar Surface: How All the Parts Fit Together.” *Space 94.* Wisconsin Center for Space Automation and Robotics Technical Report (WCSAR-TR)AR3-8808-1
- Gajda, M. (2006). “A Lunar Volatiles Miner.” University of Wisconsin-Madison, M.S. Thesis.
- Olson, A. D. S. (2013). “The Mark IV: A Scalable Lunar Miner Prototype.” *International Astronautical Congress 2013, Beijing, China, IAC-13.A3.2B.7*.
- Olson, A. D. S., Santarius, J. F., and Kulcinski, G. L. (2014). “Design of a Lunar Solar Wind Volatiles Extraction System.” *AIAA SPACE 2014 Conference and Exposition*.

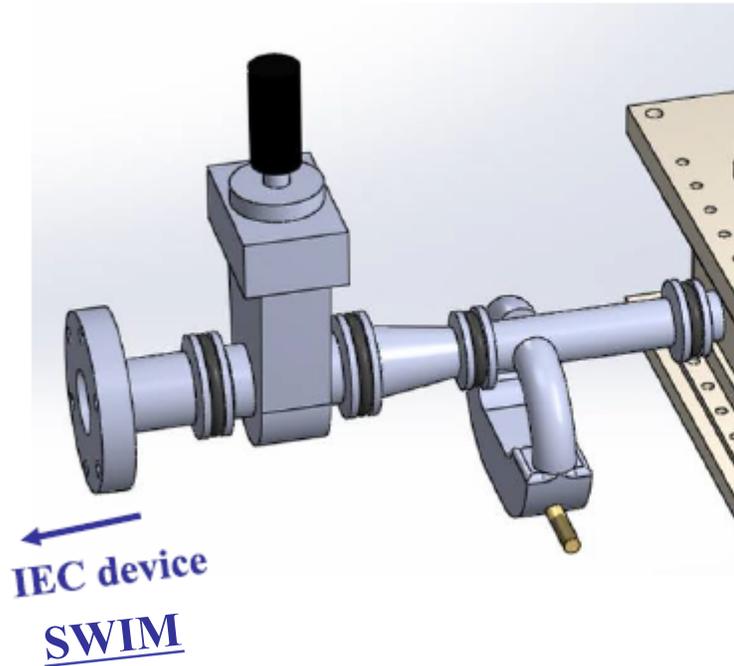
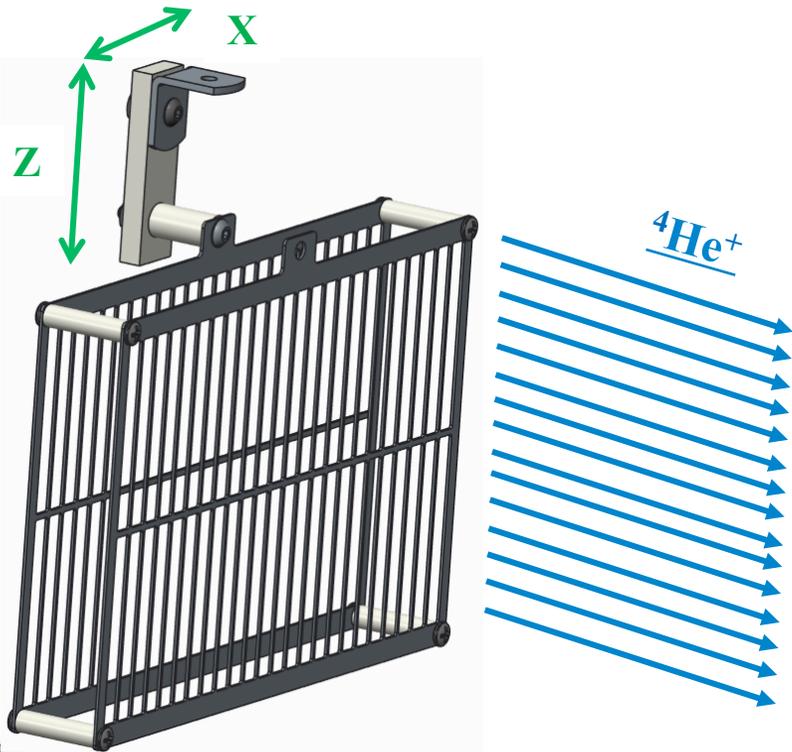


Backup/Reference Slides

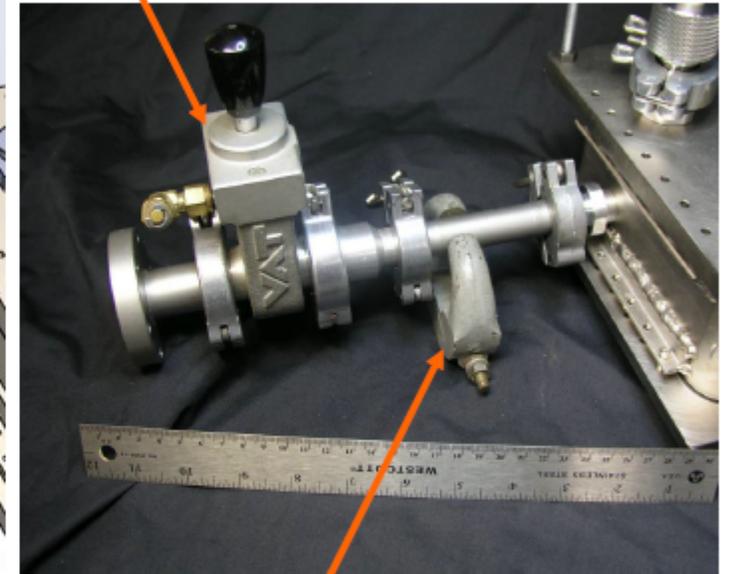
Helium Ion Energy Testing

Energy spectrum testing concept:

- Attach particle energy analyzer to SWIM chamber
- Mount SWIM grid assembly onto a moveable (2 DOF) platform
- Measure ${}^4\text{He}^+$ energy: vary pressure, grid voltage, and filament settings



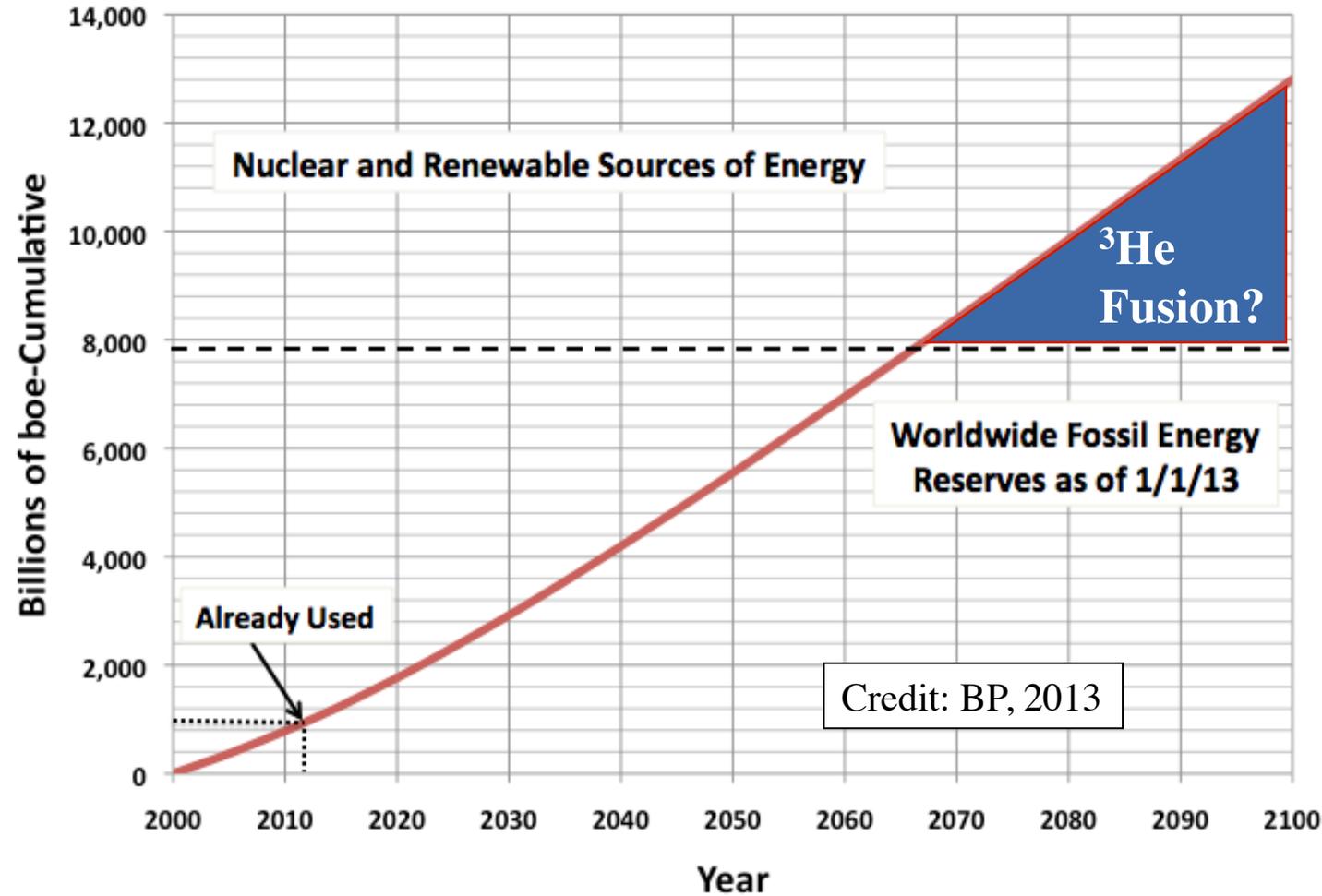
Gate valve



Bending magnet

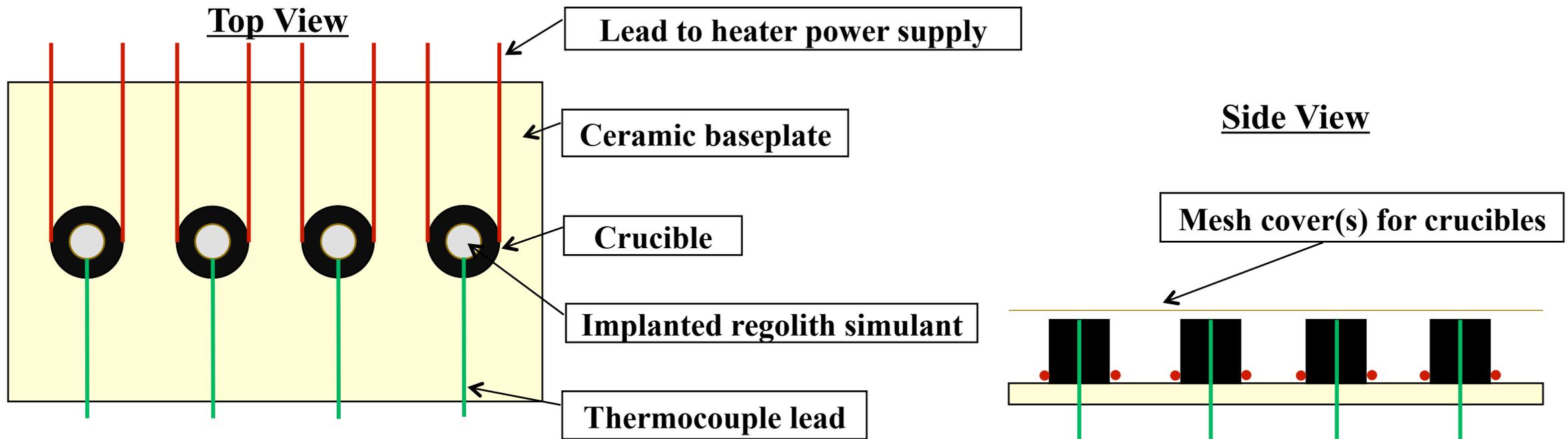
^3He Fusion Can Help Satisfy Future Energy Needs

- Energy usage is increasing
- Nuclear and renewable energy will be key in meeting our future needs
- Helium-3 fusion can be a part of the energy solution and enables unique space applications



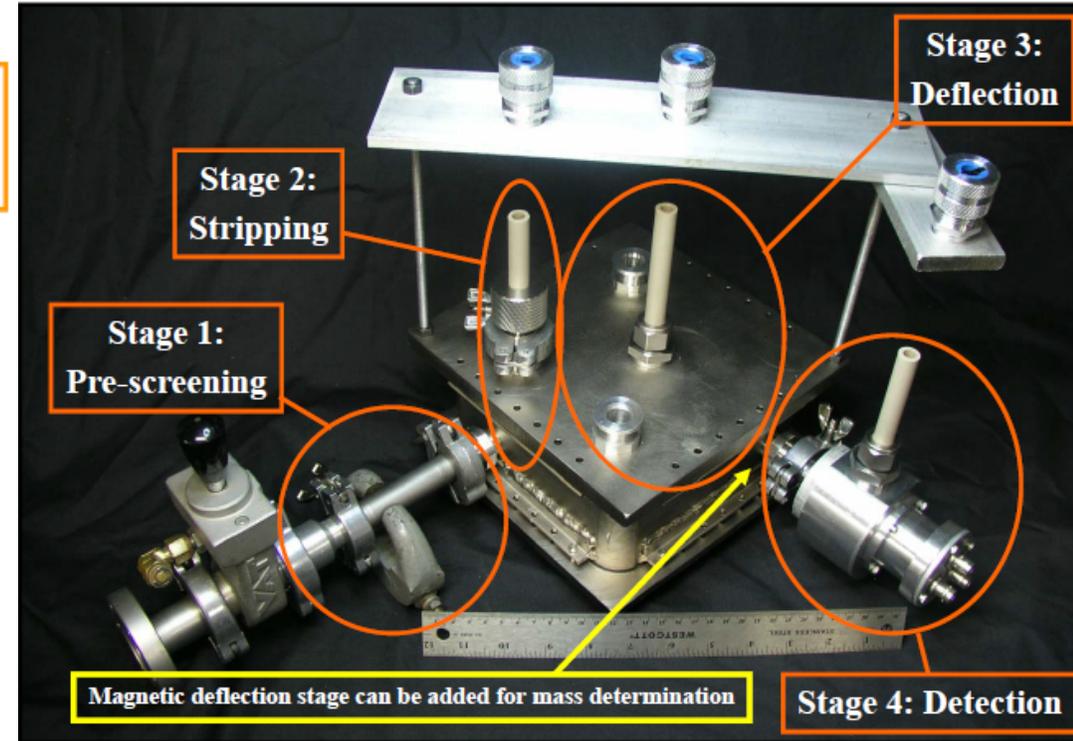
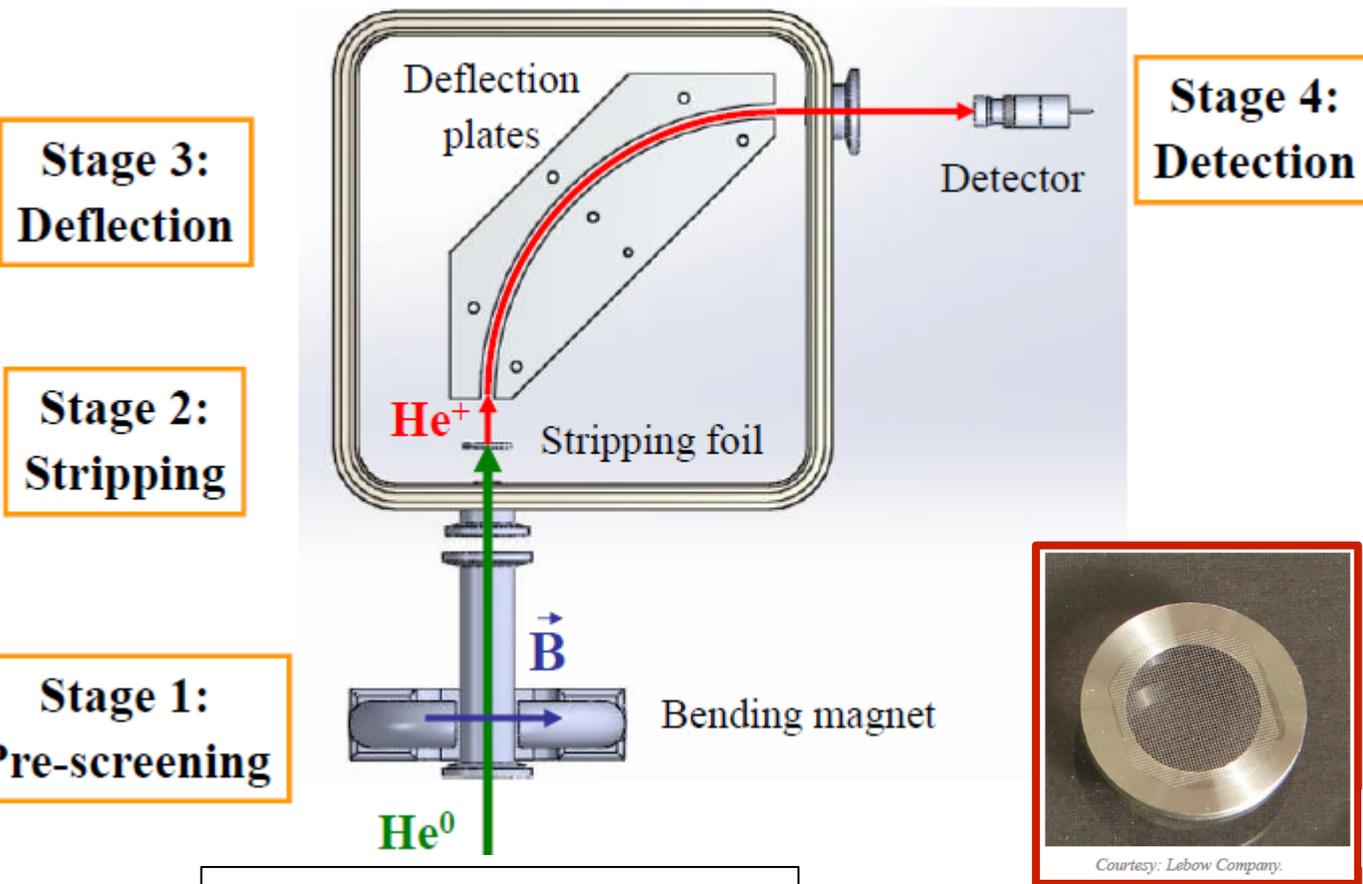
Sample Concentration Analysis (SCAN) Concept

- Within a vacuum system with a turbo pump and RGA
- Ceramic crucibles with specialized insulating heaters and a ceramic baseplate
- Heat implanted simulant in the crucibles to $\sim 1000^{\circ}\text{C}$ (step-wise heating)
- Use thermocouples to record regolith temperature



Neutral Particle Analyzer

Use the NPA as an Ion Energy Diagnostic: Remove stripping/screening foil



Thin Carbon foil (~10nm thickness)

Credit: Becerra et. al, 2014



System Monitor and Controller

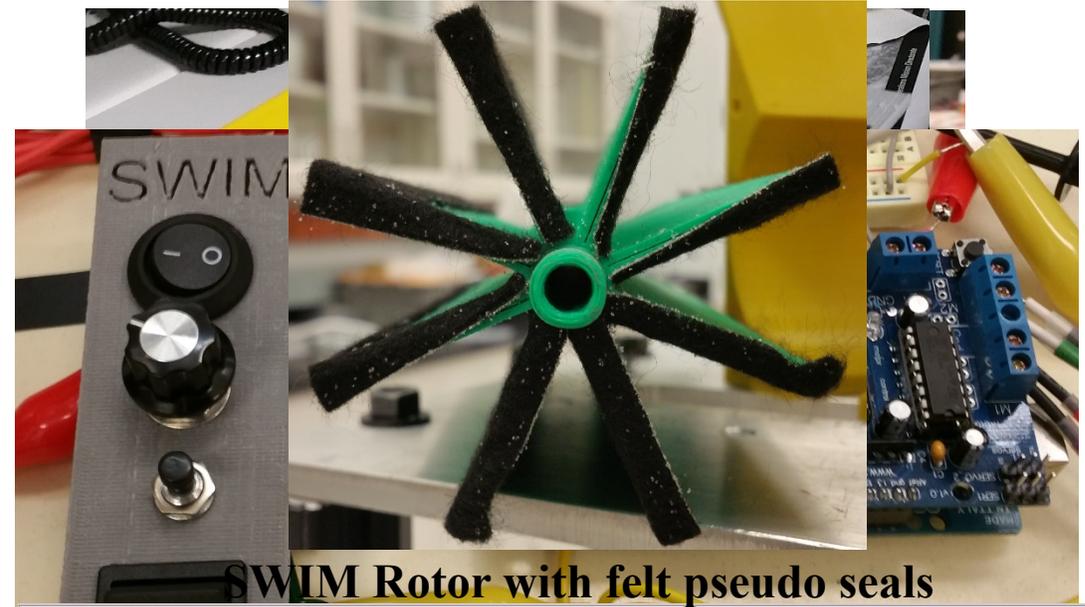
	Input	Connection
Pressure	Chamber Convectron	BNC-1
	Chamber Ion	BNC-2
	NPA TC	BNC-3
	NPA Ion	BNC-4
Grid High Voltage	Voltage Monitor	DB25-E
	Current Monitor	DB25-E
NPA High Voltage	Voltage Monitor (3)	DB25-B,C,D
		DB25-B,C,D
NPA Measurement	Energy Signal	BNC-5
Flow Controller PS	-15 V	BNC-6
	+15 V	BNC-7
	Flow Signal	DB9-A
Filament PS	Filament Power Supply	DB25-A
HV Interlock	High/Low Signal	DB25-A
Motor PS	Motor Power Supply	Barrel Jack

	Output	Connection
2D Mount	(8) Leads	DB25-A
Filament PS	(2) Leads	DB25-A
Grid HV PS	Voltage Control	DB25-E
NPA HV PS	Voltage Control (3)	DB25-B,C,D
Flow Controller	Control Signal	DB9-A
CPU	LabVIEW	USB-1

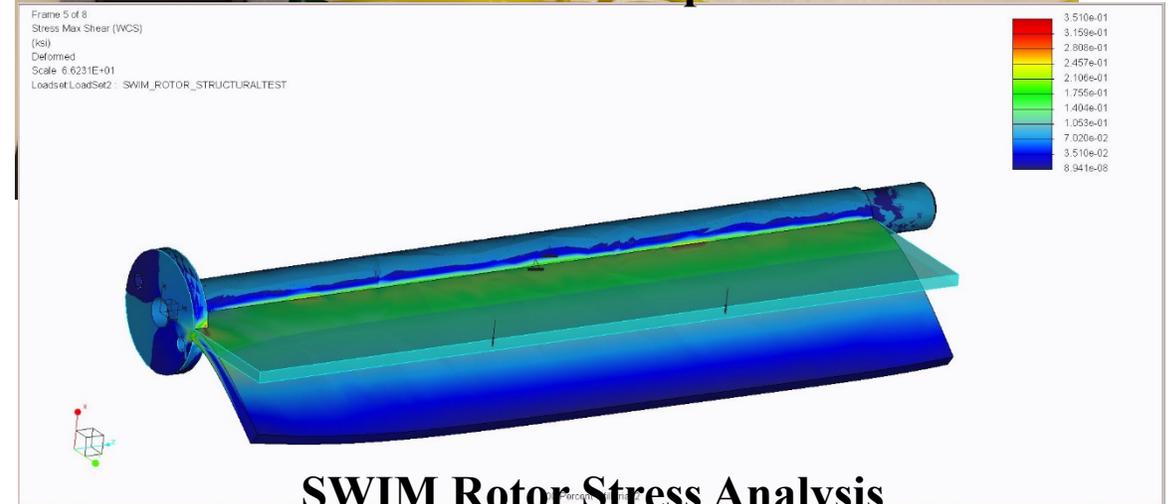
- (11) Analog Inputs
- (5) Analog Outputs
- (1) Digital Inputs
- (0) Digital Outputs
- (9) BNC Ports
- (5) DB25 Ports
- (1) DB9 Port
- (1) USB Type B Port

SWIM: Hopper/Feeder

- Almost completely 3D printed PLA plastic
- Rotor made thick enough for motor torque
- Inside surfaces lined with adhesive backed aluminum to bleed off static charge
- Arduino Mega (Atmel 2560) microcontroller and Adafruit motor shield for rotor control



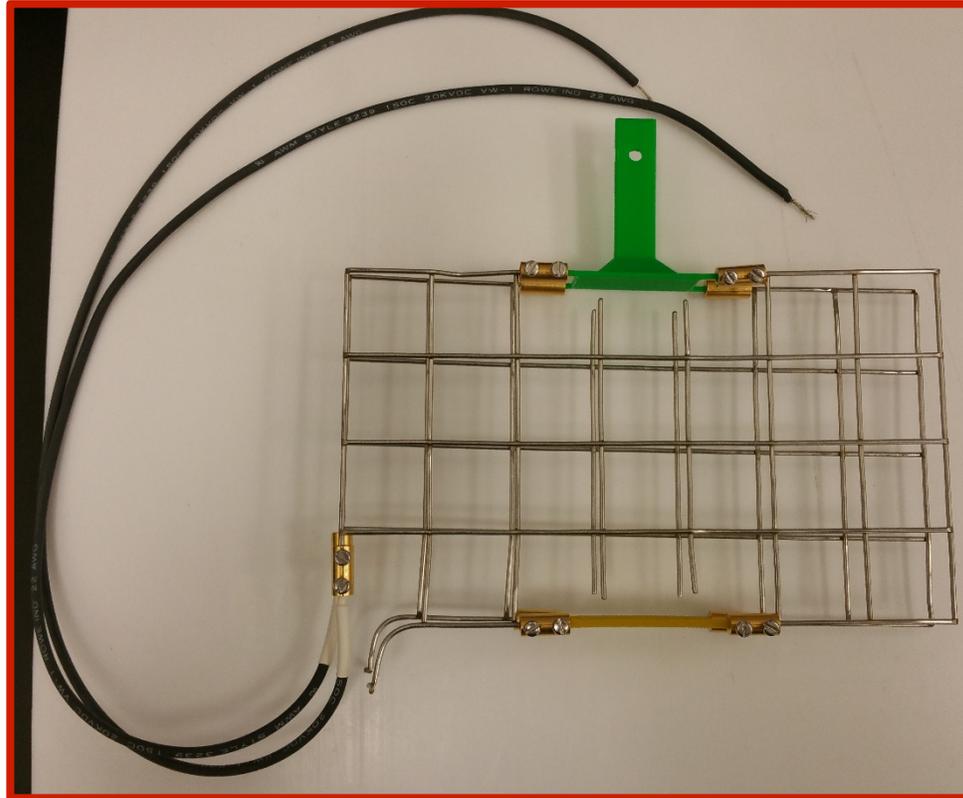
SWIM Rotor with felt pseudo seals



SWIM Rotor Stress Analysis

SWIM: Ion Acceleration Grids

- 90% Open Stainless Steel Welded Wire Mesh (1/16" diameter wire)
- 5/8" grid separation
- PLA Support struts (stable to ~60 °C)
- Brass wire connectors
- High voltage wire (up to 30kV, 150 °C)



SWIM Grid Assembly



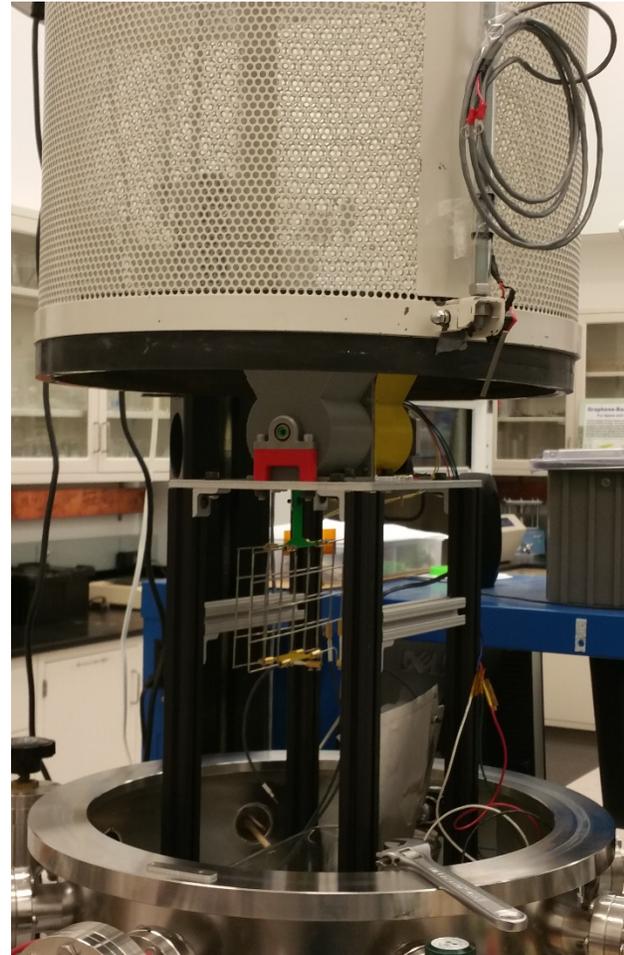
SWIM Grid Installed

SWIM: Vacuum Chamber at NASA KSC

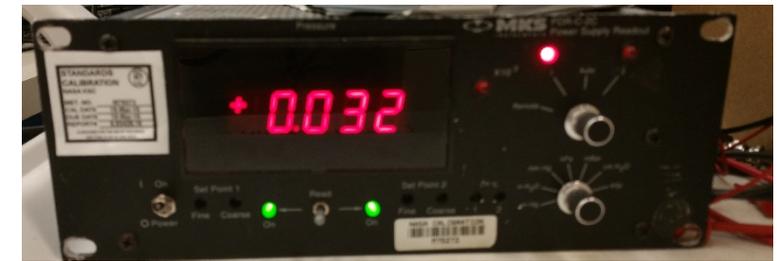
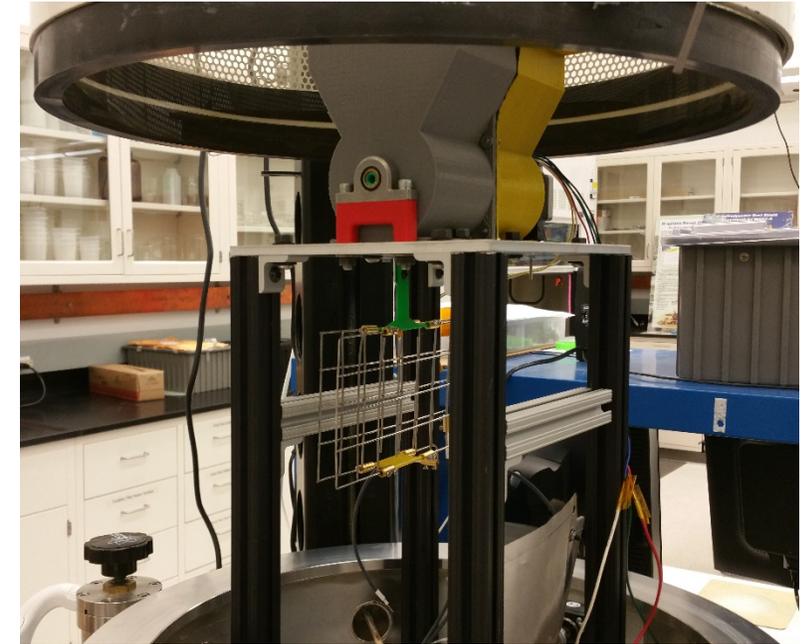
- Bell jar vacuum system in Electrostatics and Surface Physics Lab (ESPL)



SWIM System in ESPL

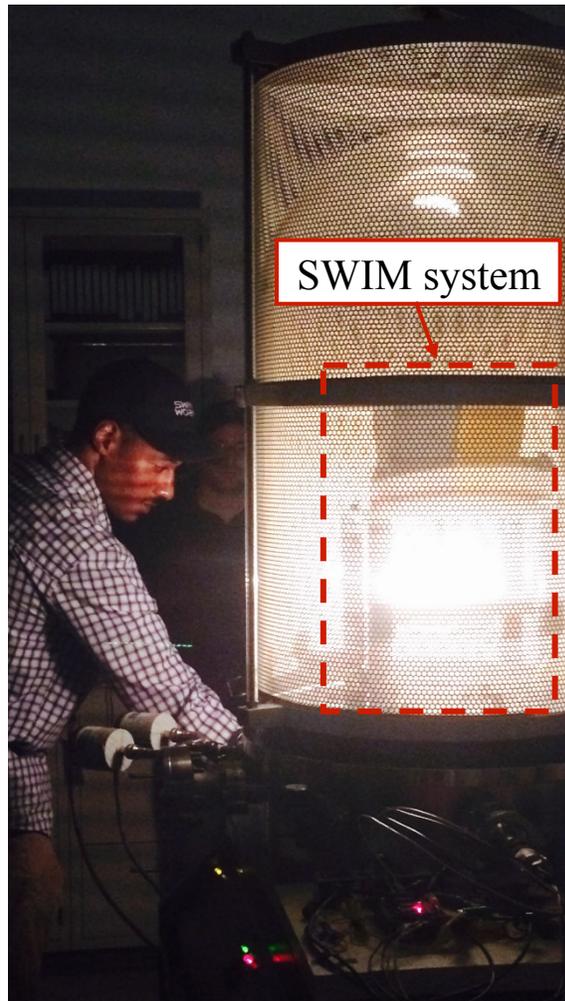


SWIM Vacuum System Open

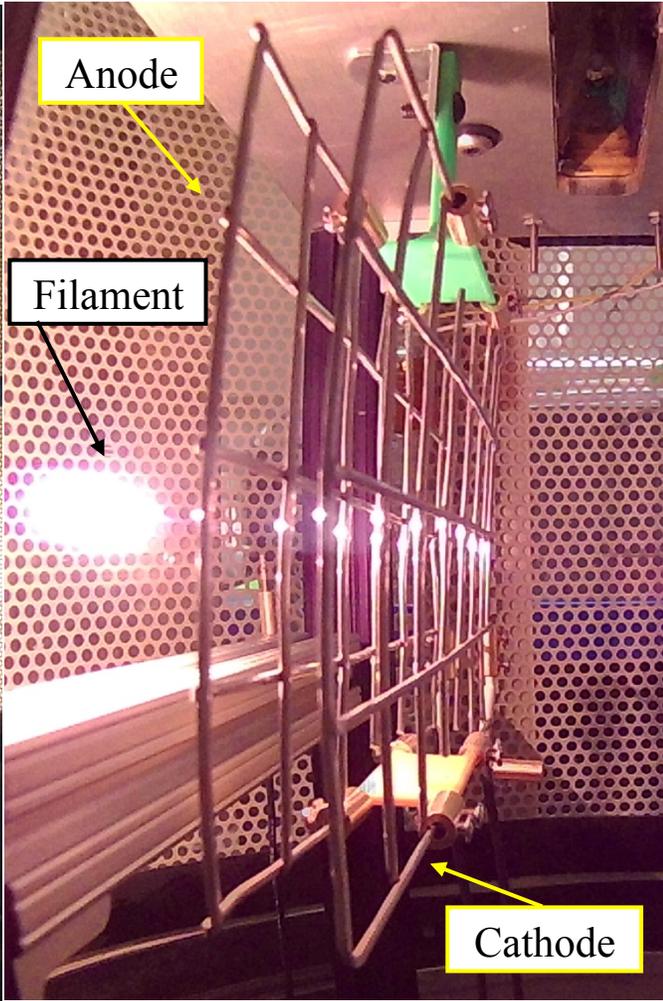


Pressure Monitor

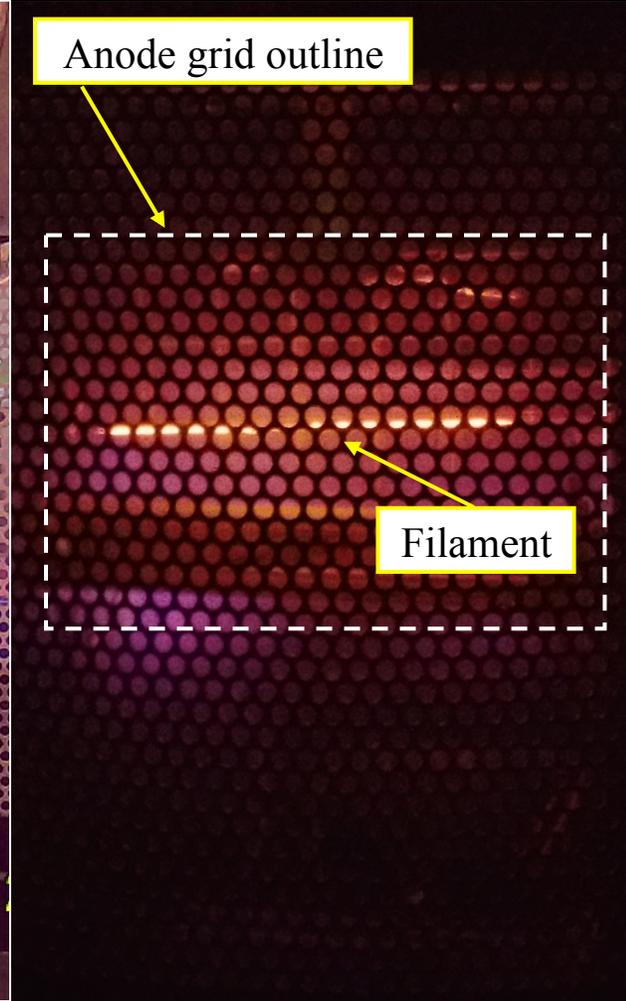
SWIM: Device Operation



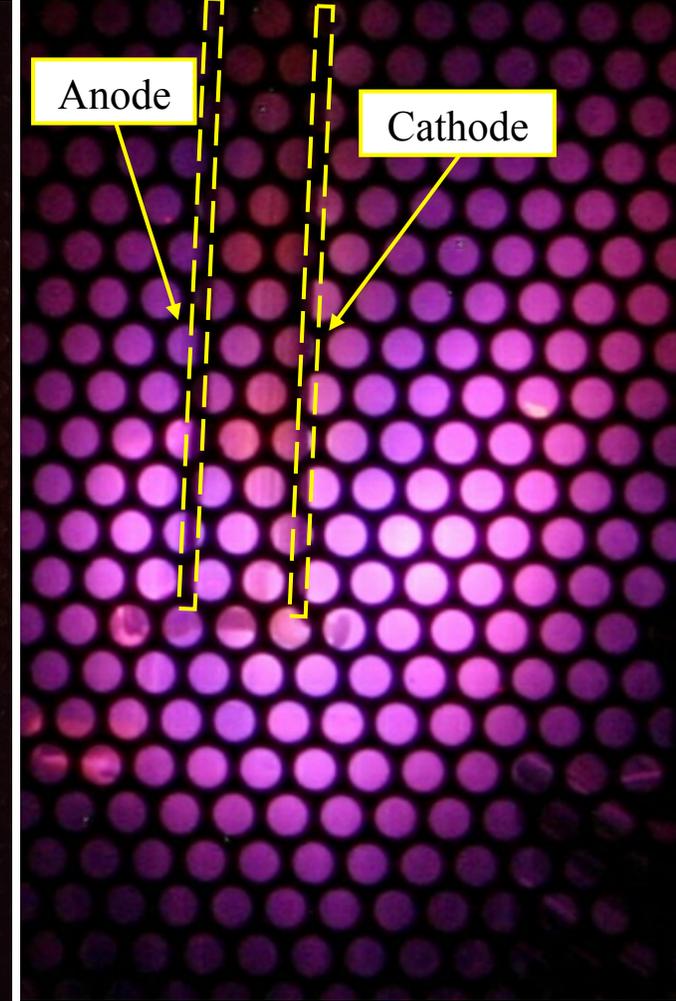
Filament on only



Action cam image



Filament on and discharge
(side view)



Discharge (front view)

SWIM: Issues from Initial Device Operation

- Grids
 - Eliminate “sharp” points
 - Increase the ratio of the inter-grid spacing to mesh gap width
 - Potentially ceramic support pieces instead of PLA
- Filaments:
 - Increase bias voltage to $\sim 200\text{V}$ range
- Instrumentation
 - Current measurement: LabView VI and DAQ (or microcontroller)
- Camera Issues
 - More reliable method needed



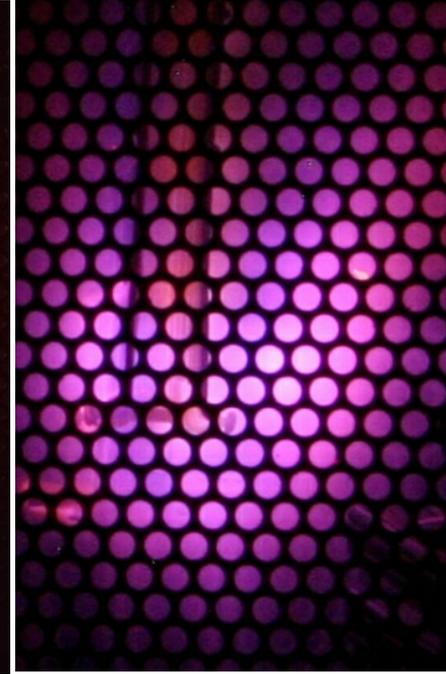
Filament on only



Action cam image



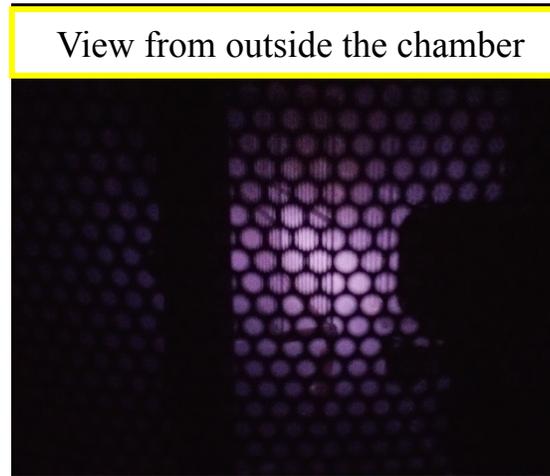
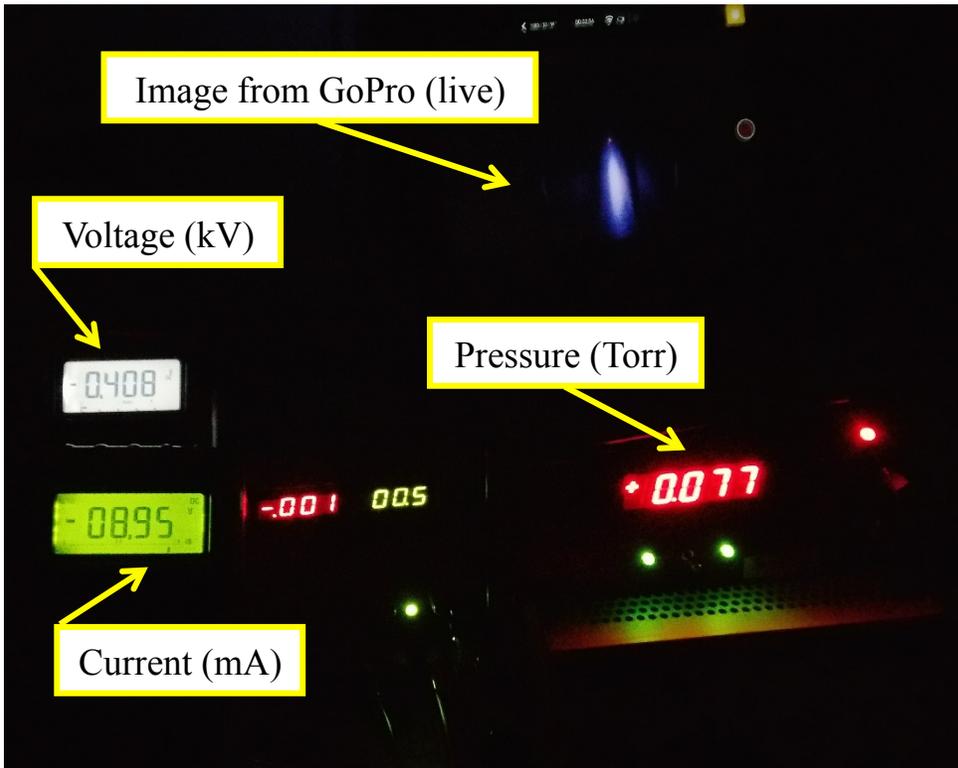
Filament on and discharge (side view)



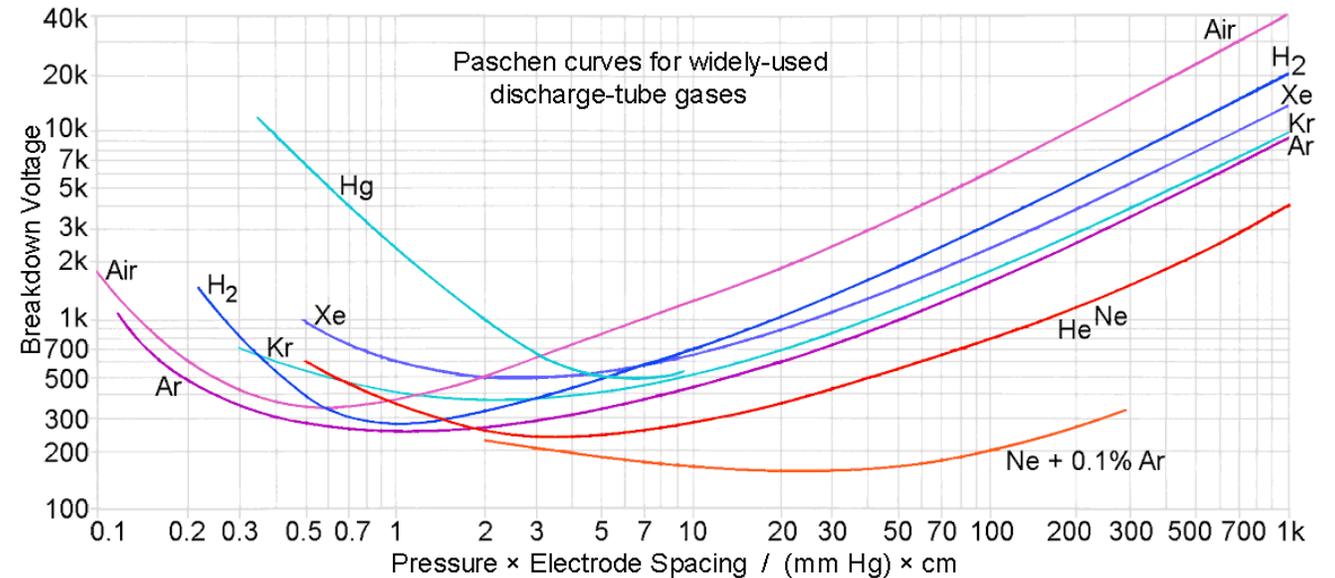
Discharge (front view)



SWIM: Tests with New Grids

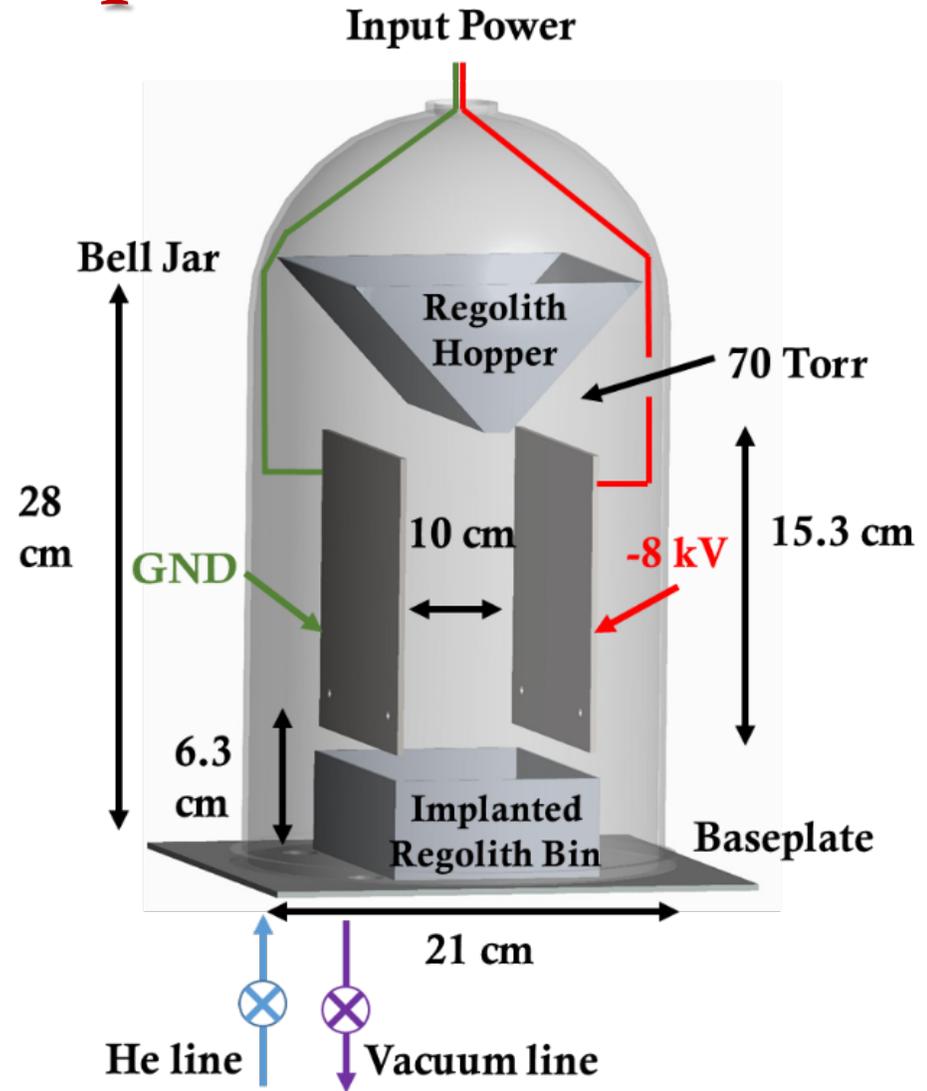


Discharge occurs approximately where the Paschen curve suggests for the same pressure-distance product of 0.2 Torr-cm at ~400V for **air**



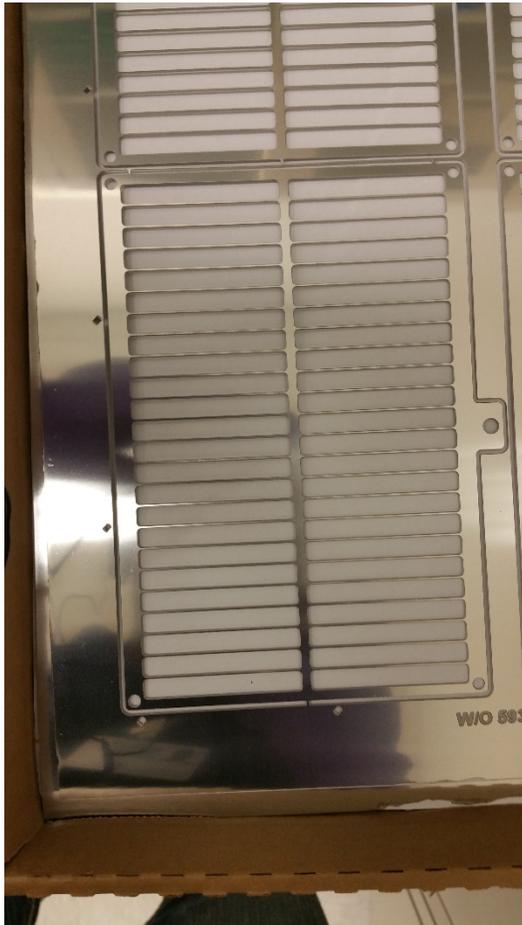
SWIM Initial Concept

- Drop simulant downward between a dc parallel electrode discharge
- Cathode held $\sim 8\text{kV}$ below anode to accelerate positively charged helium ions toward the cathode
- Accelerated ions moving between the electrodes impact the falling simulant particles and are thus implanted
- Process done in a bell jar vacuum chamber filled with low pressure helium-4



Original SWIM Concept

SWIM: New Grids from Fotofab, Inc.



Arrived 12/10/2015 at KSC



2 sets of (4) 1/32" thick 316 stainless steel grids (\$750 for 8 grids)



Testing in Minnesota

- Dr. Pepin can test samples of helium implanted simulant in the degassing system in the Physics dept. at the University of Minnesota
- Sample requirements
 - Wrapped in 5mm x 7mm Pt foil (provided by the lab)
 - Heated up to 1400°C by direct resistance heating
 - ~1 hr/test
 - ~10 samples/week can be tested
 - There are samples already in line for testing
 - Lab will close in mid July at the latest

SWIM Sampling Strategy

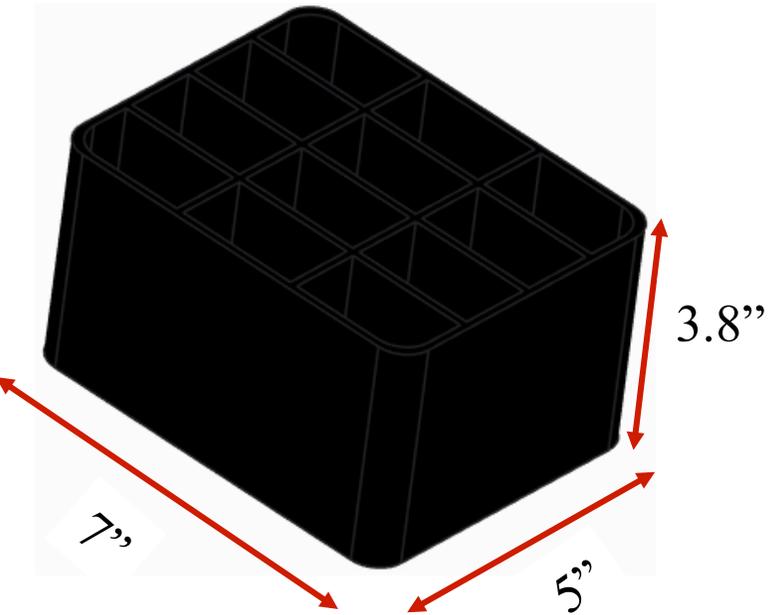
■ Objectives

- Obtain a helium concentration vs. (helium flux/simulant mass flow rate) curve
- Obtain some info the variability of helium dose into the simulant stream

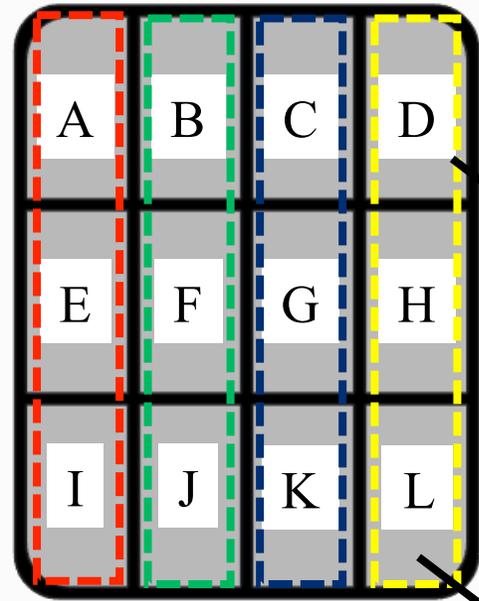
■ Setup

- Load hopper with 240 g of <100 micron JSC-1A simulant
 - Each rotor scoop holds ~60 g of simulant
- Drop rotor scoops at 4 different helium flux levels corresponding to an estimated 10, 20, 40, 80 ppb concentration into a 12 section partitioned bin
- Take three ~15 mg samples per section at random for 36 total samples
 - Sample foil enclosures can hold ~15 mg of <100 micron JSC-1A

SWIM Sample Bin

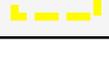


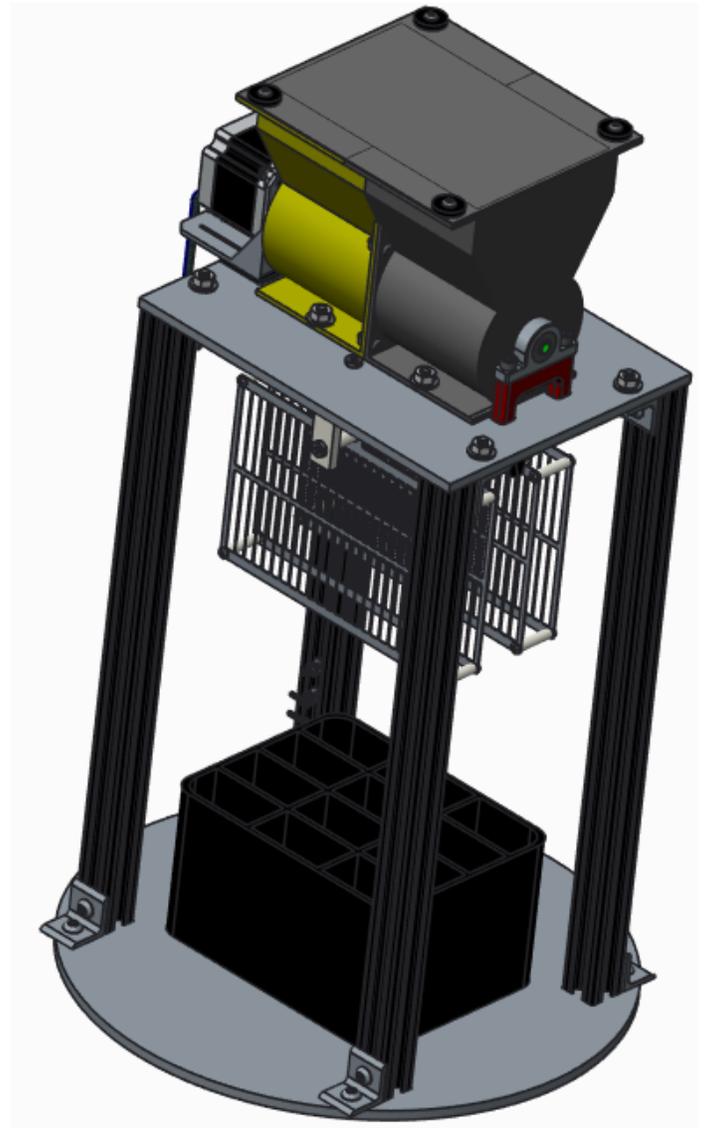
60 g per
implantation level



20 g/section

Three 15 mg
samples

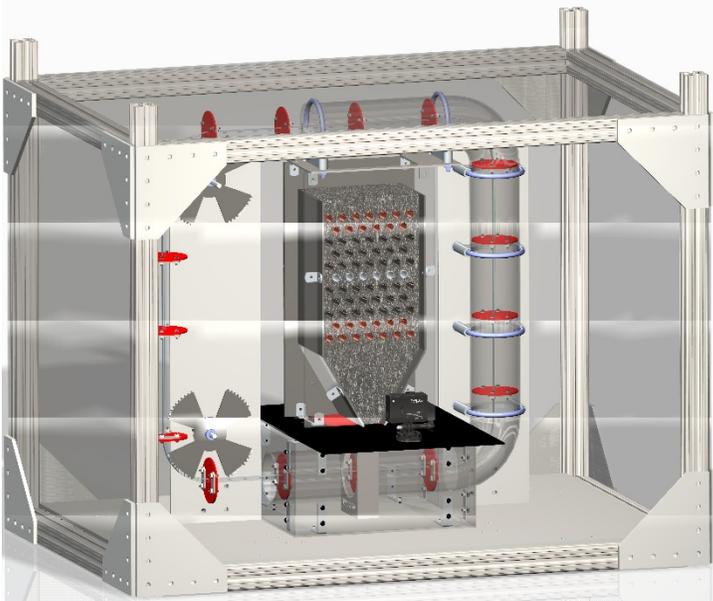
	10 ppb
	20 ppb
	40 ppb
	80 ppb



Follow-On Wisconsin ^3He Extraction Research Roadmap

Lab Prototypes

- Heating System Optimization
 - ^3He Release Measurement



(UW HEAT GFX)

Lunar Gravity Testing

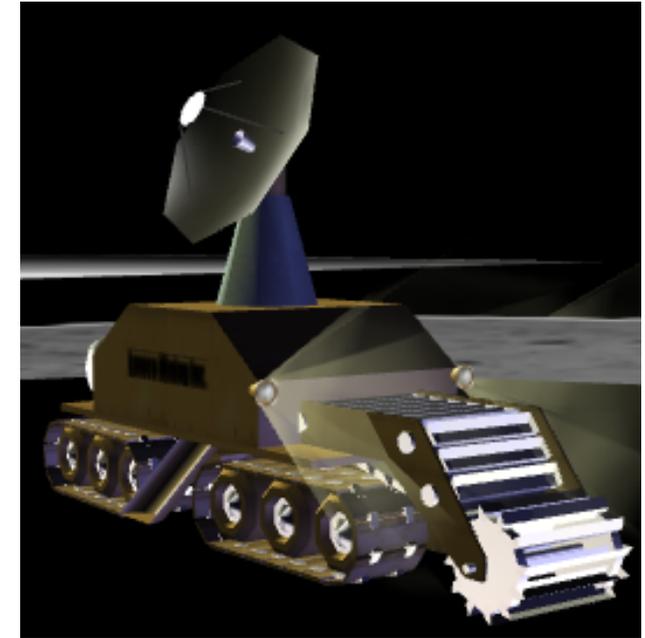
- Regolith Flow through Heating System in Lunar Gravity



(Lunar G HEAT)

Small Scale Lunar Mission

- INTERLUNE 1 Concept
- Use of GLXP Provider(s)?



(Micro ^3He Miner)



The ^3He on Earth is Insufficient to Support Fusion Power

Three sources of helium-3 on the Earth

- **Atmosphere:**
 - helium-3 concentration in atmosphere is $\approx 7 \times 10^{-12}$ by volume.
 - total amount in the entire atmosphere is $\approx 4,000$ tonnes
- **Natural Gas:** potentially as much as 280 kg in reserves and speculative sources that are not being tapped
- **Decay of Tritium:** tritium decays into helium-3 with a 12.3 year half life and 2-4 kg/yr of helium-3 is produced from tritium in the U.S. and Canada

Shortage

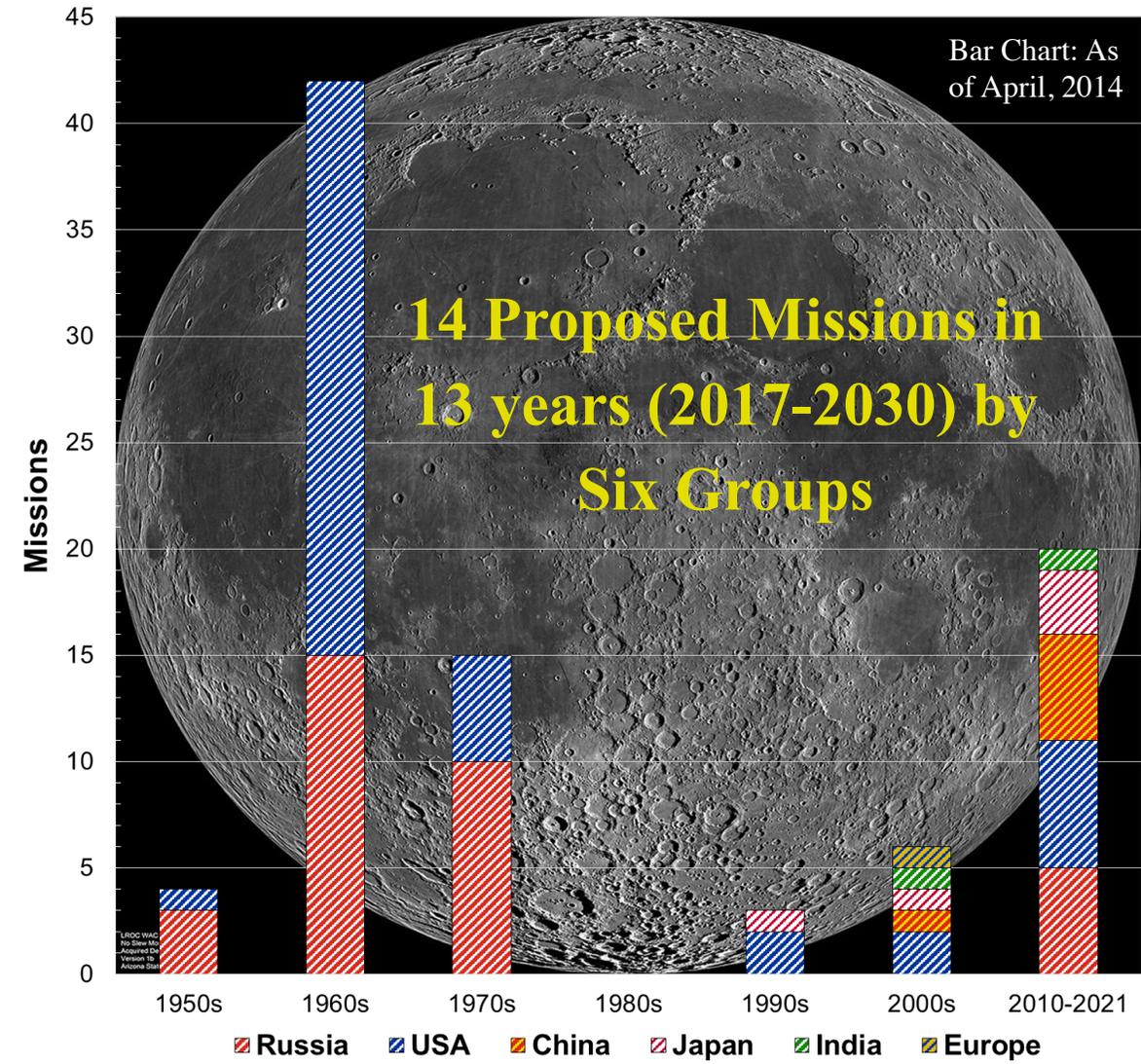
- **Increased Usage:** supply depleted of helium-3 from tritium decay (down to < 10 kg in the U.S. as of 2010)
- **Increased Price:** from ($\sim \$1,000,000/\text{kg}$) to ($> \$30,000,000/\text{kg}$)
- **Available for Fusion R&D:** only ~ 10 kg ^3He (200 MW-y fusion energy) is accessible on Earth



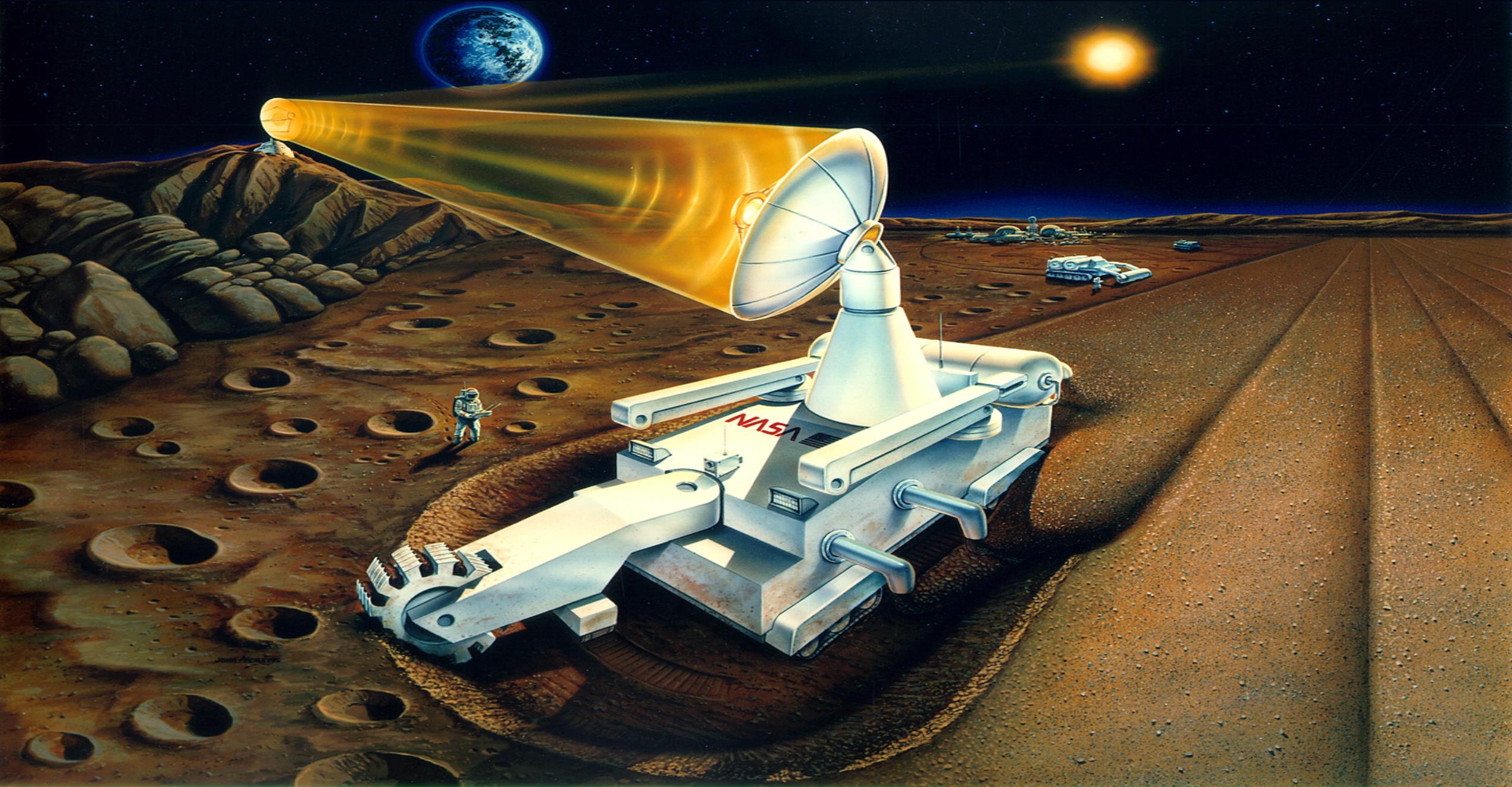
Interest in Lunar Exploration is Ramping Up

Lunar Mission Manifest

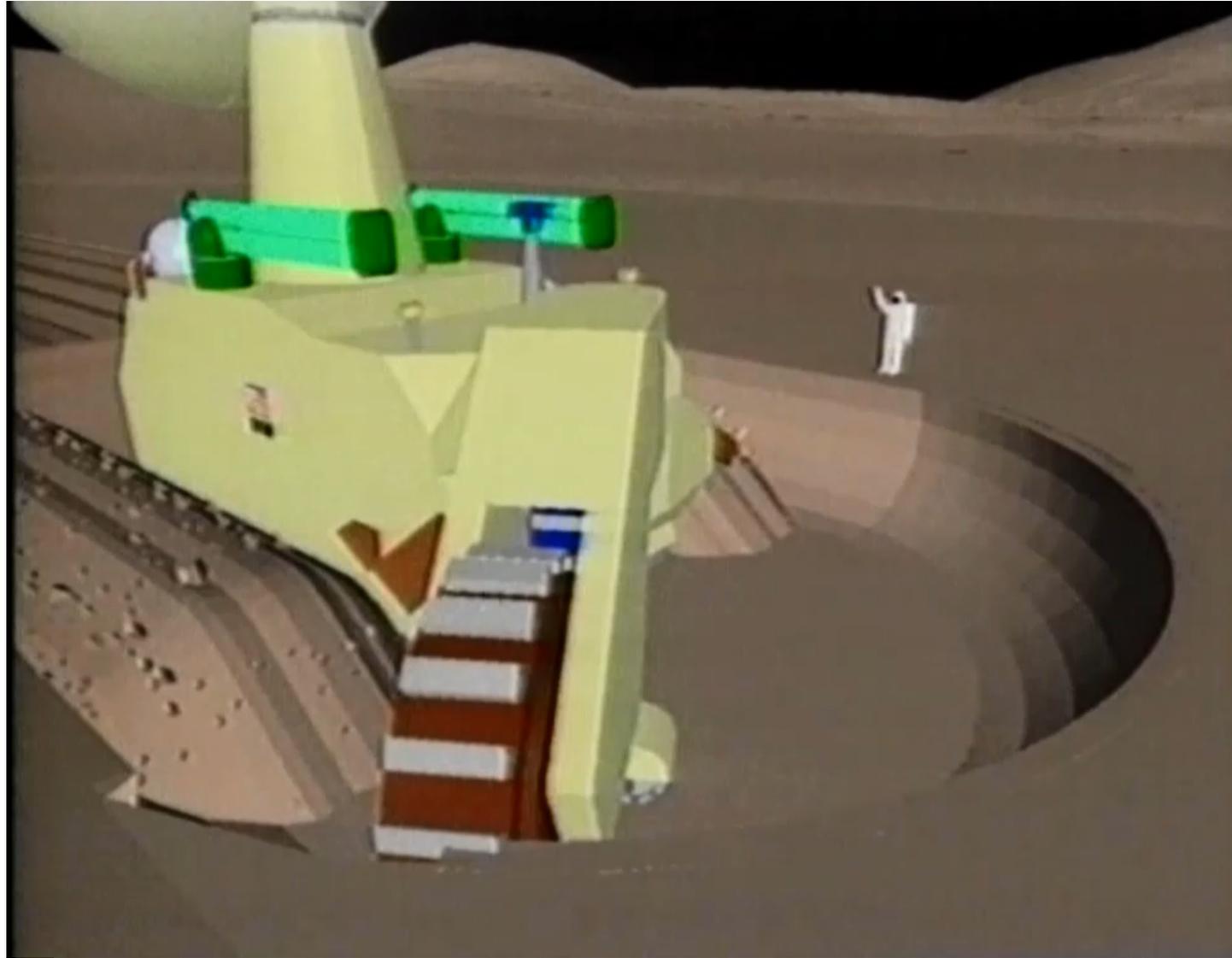
Launch Date	Organization	Country	Mission
2015-2018	Google Lunar XPrize (GLXP)		GLXP
2017	CNSA	China 	Chang'e-5
2017	JAXA	Japan 	SELENE-2
2018	ISRO	India 	Chandrayaan-2
2018	NASA	USA 	RP
2018	NASA	USA 	EM-1
2020	CNSA	China 	Chang'e-6
2021	NASA	USA 	EM-2
2020s	Roscosmos	Russia 	Luna-25
		Russia 	Luna-26
		Russia 	Luna-27
		Russia 	Luna-28
		Russia 	Luna-29
JAXA	Japan 	SELENE-3	



Lunar Helium-3 for Clean Fusion Power on Earth or in Space



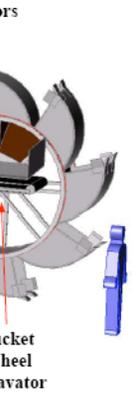
There are Three Main Processes in a ^3He Miner



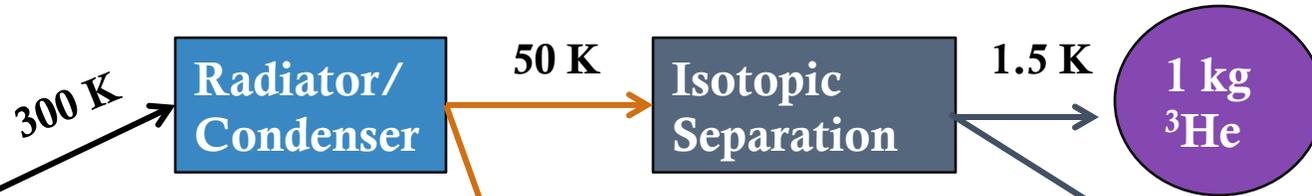
There are Three Wisconsin Lunar ^3He Miner Designs

Excavation rate (tonnes/hr)	1258
Processing rate (tonnes/hr)	5563
Thermal processing goal (tonnes)	14,200
^3He extraction (kg/yr)	66

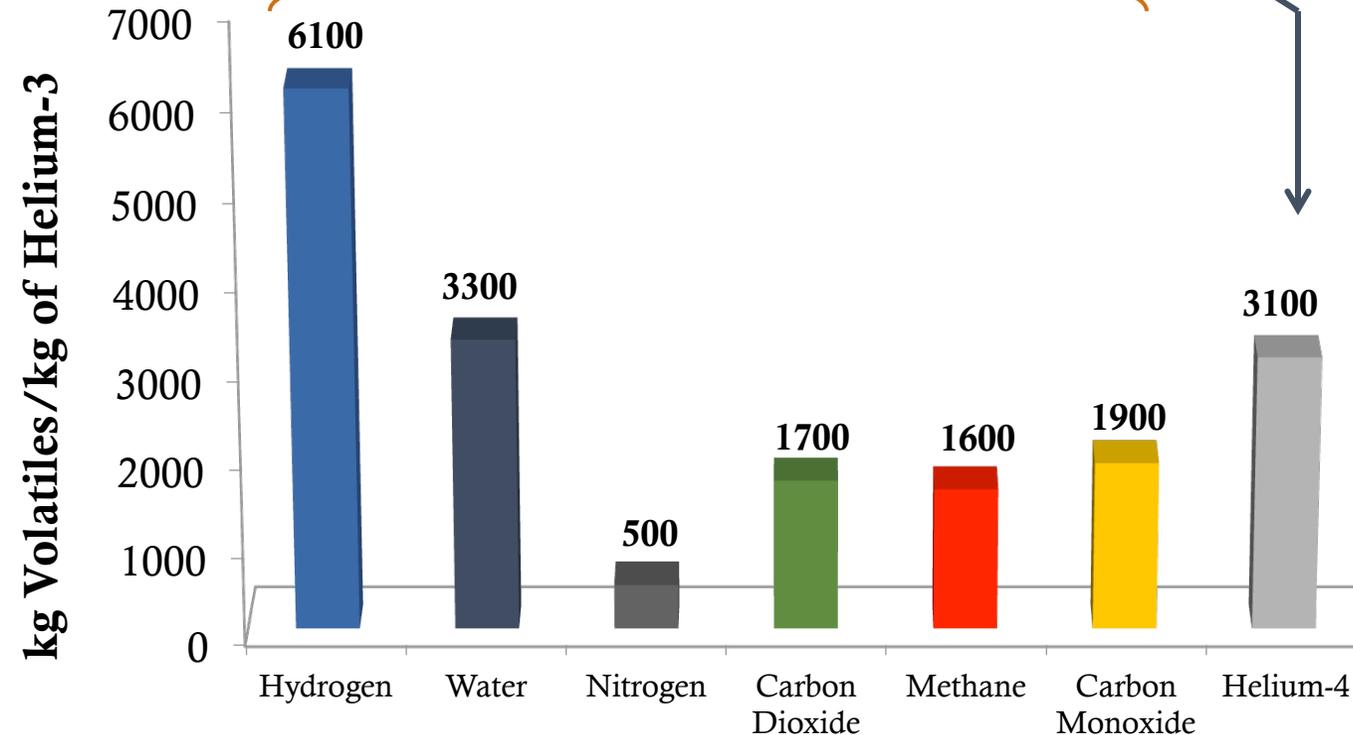
TAKRAF SR 8000 bucket-wheel excavator can excavate > 16,000 tonnes/hr of bituminous coal and has a mass of 14,200 tonnes



³He Mining Produces Valuable ISRU Volatile By-Products

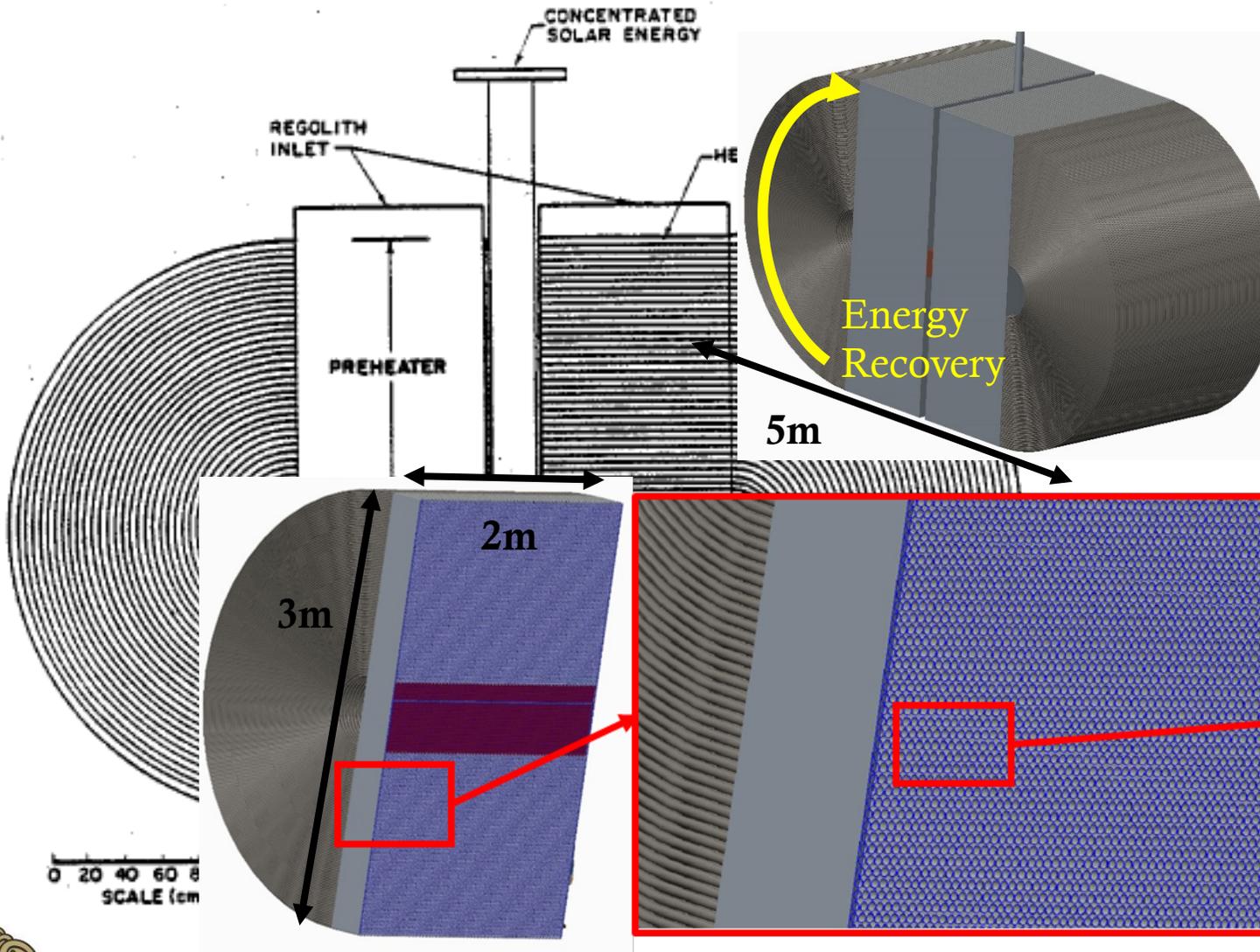


- Fuel: H₂, CH₄, H₂O
- Life support:
CO₂, CH₄, N₂, H₂O
- Cryogenics: ⁴He



Results from Apollo 11 sample 10086.16

The Heat Pipe Heat Exchanger is a Key Part of the Miner



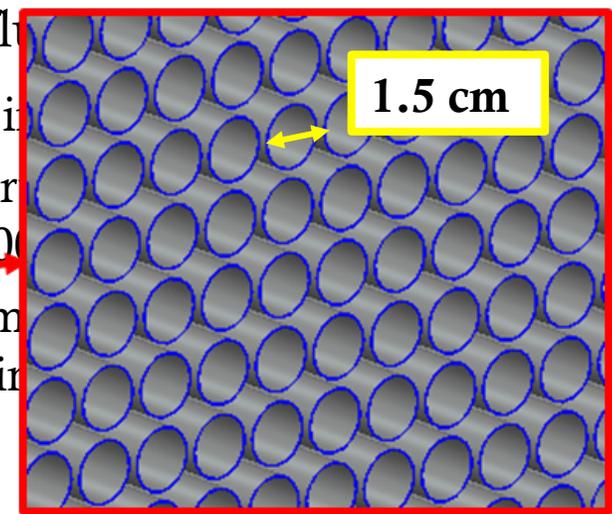
energy recovery by using heat pipes

170 MW from solar collector (70 MW reduction)

157 kg/s of regolith from 30 °C up to 700 °C
 release 85% of embedded ³He

releases 16.7 g/hr of ³He (66 kg in 3942 hours of
 operation)

combinations
 to ~250 °C
 operating between
 stainless steel pipes:



JSC-1A Lunar Regolith Simulant Will be Used



Minerals	JSC-1A
Plagioclase	37.83
Clinopyroxene	18.77
Orthopyroxene	0.66
Olivine	12.44
Glass	26.67
Magnetite	0.01
Chromite	0.00
Ilmenite	0.11
Sulphides	0.17
Iron	0.00
MgFeAl Silicate	3.06
K Feldspar	0.07
Quartz	0.01
Calcite	0.11
Others	0.07
Total	100.00

Der
Me
Me

Thanks to Dr. James Manton for 101 kg of JSC-1A from NASA JSC

Specific heat	700- 1400 [J/kg-K]	1047.41 log(Temp) - 1848.15 [J/kg-K]
Thermal Conductivity @20 kPa	0.15 - 0.40 [W/m-K]	3.9e ⁻⁴ (Temp) + 0.1588 [W/m-K]
Cohesion	1 kPa	1 kPa
Angle of Internal Friction	40-55 degrees	45 degrees

JSC-1A Merriam Crater Cinders (left) and ground regolith simulant (right). Credit: ORBITEC

The ^3He on Earth is Insufficient to Support Fusion Power

Three sources of ^3He on the Earth

- **Atmosphere:**
 - helium-3 concentration in atmosphere is $\approx 7 \times 10^{-12}$ by volume.
 - total amount in the entire atmosphere is $\approx 4,000$ tonnes
- **Natural Gas:** potentially as much as 280 kg in reserves and speculative sources that are not being tapped
- **Decay of Tritium:** tritium decays into helium-3 with a 12.3 year half life and 2-4 kg/yr of helium-3 is produced from tritium in the U.S. and Canada

There is enough ^3He on Earth for Fusion R&D

- **Available for Fusion R&D:** ~ 100 kg ^3He (2000 MW-y fusion energy) is accessible on Earth

A Potential Roadmap for Lunar Helium-3

Prospecting and Research ~2015-2025

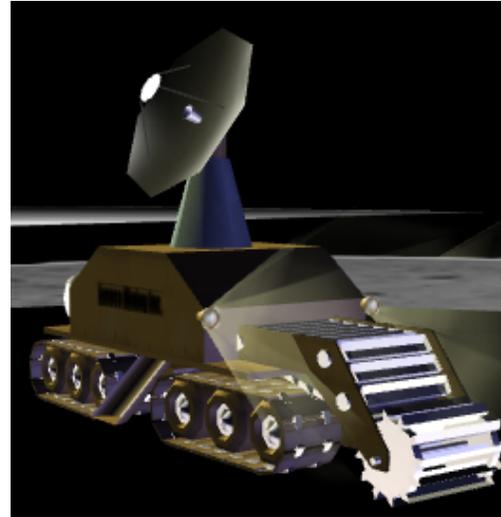
- Government missions
- **Research at UW-Madison**
- Google Lunar X-Prize



(NASA RESOLVE)

Small Scale Mining ~2025-2035

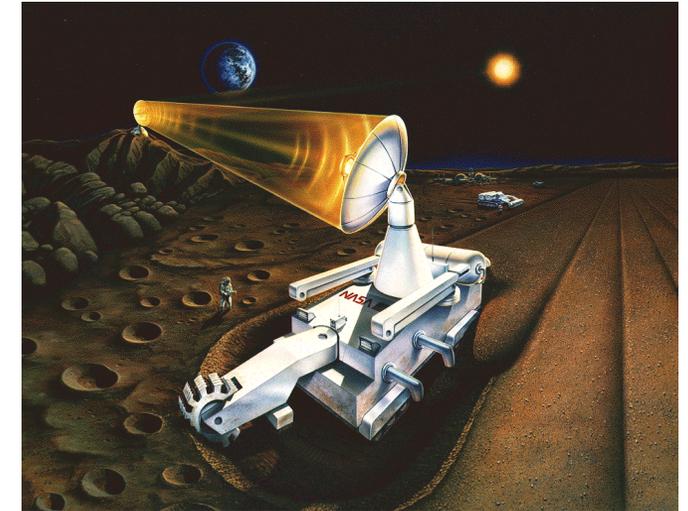
- First small test lunar miners
- Start of lunar outposts
- Other mining activities starting



(Small scale concept)

Established Mining Industry ~Post 2035

- Large lunar miners
- Established Moon-Earth shipping
- **^3He Fusion Reactors**



(UW Mark-II concept)



Upcoming Lunar ^3He Fusion Book

Lunar Helium-3 Fusion: The Earth's Energy Future

Follow-on to *Return to the Moon* by Dr. Harrison Schmitt, to be published in 2017

Discusses

- Increasing need for power worldwide
- Potential role of ^3He fusion technology
- Lunar bases and resources
- Legal regimes for lunar mining
- Economics of ^3He procurement
- Terrestrial fusion power plants
- Space applications of ^3He fusion
- Current lunar initiatives around the world



Authors: Dr. Harrison Schmitt, Dr. Gerald Kulcinski,
Dr. John Santarius, Aaron Olson



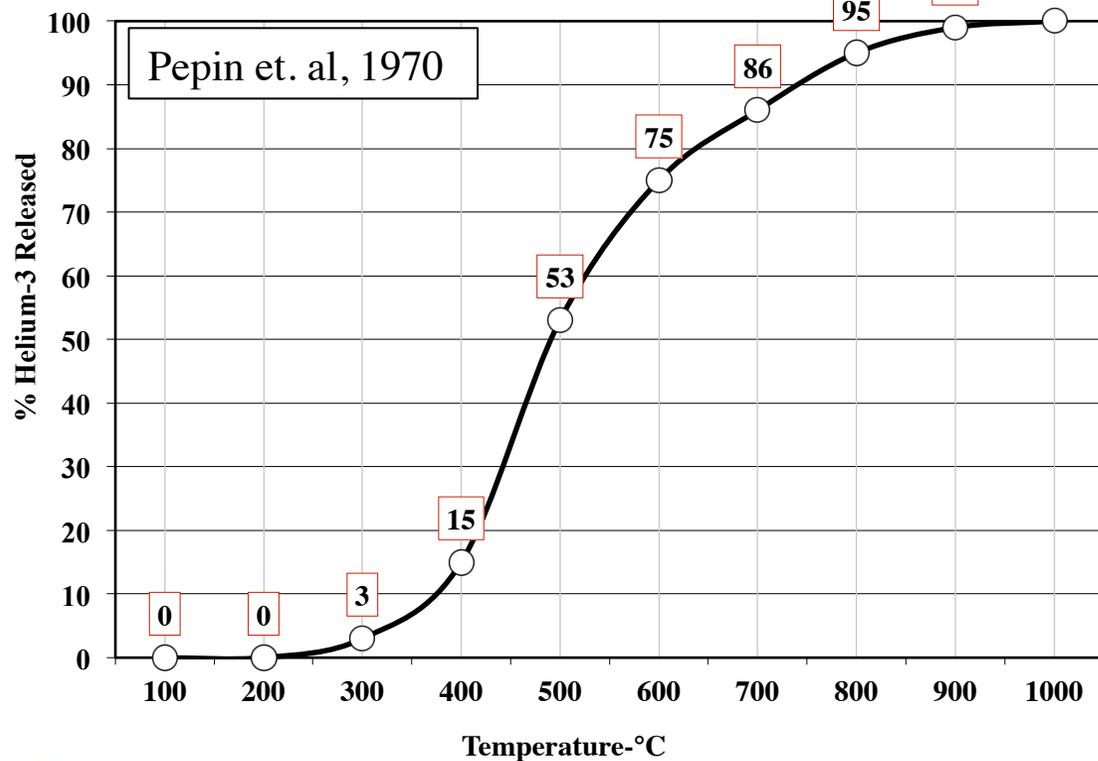
Lunar ^3He Is Well Documented

- ^3He concentration (≈ 20 ppb) verified from Apollo 11, 12, 14, 15, 16, 17 and U. S.S.R. Luna 16, 20, and 24 samples.

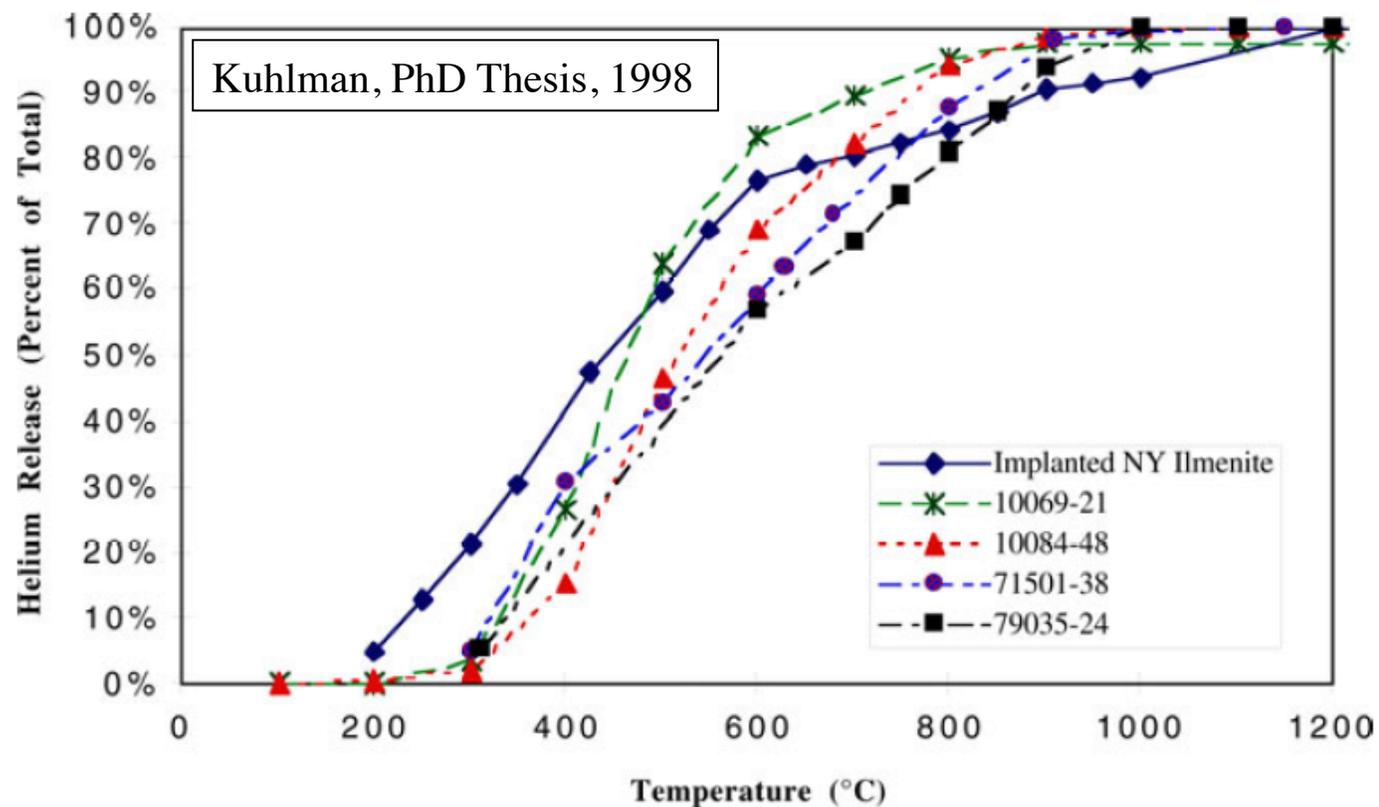
- Current analyses indicate that there are at least **1,000,000** tonnes of ^3He imbedded in the lunar soil (3m depth).

^3He Diffusion out of Lunar Regolith has been Studied

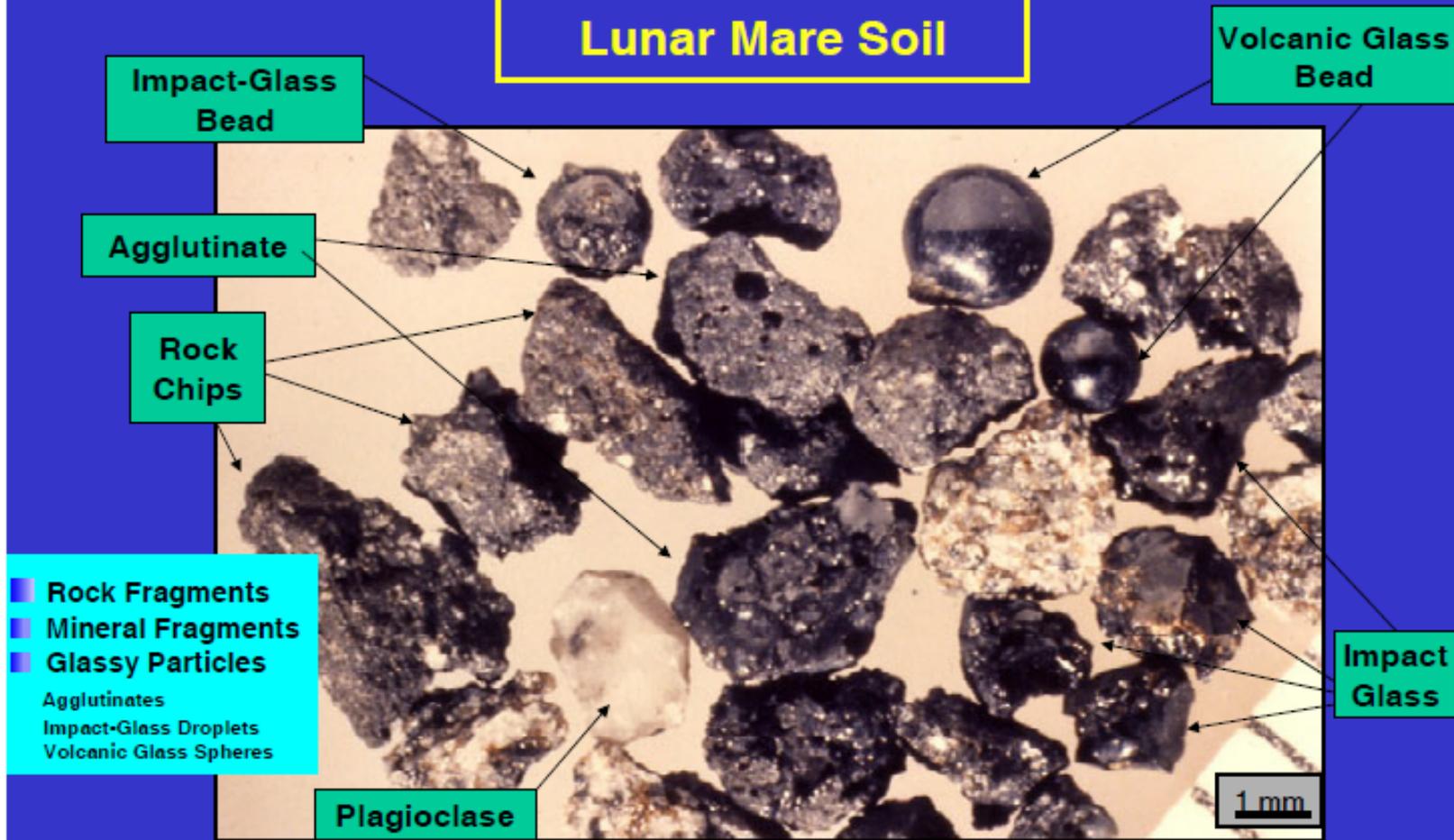
Lunar Regolith



Analog Regolith



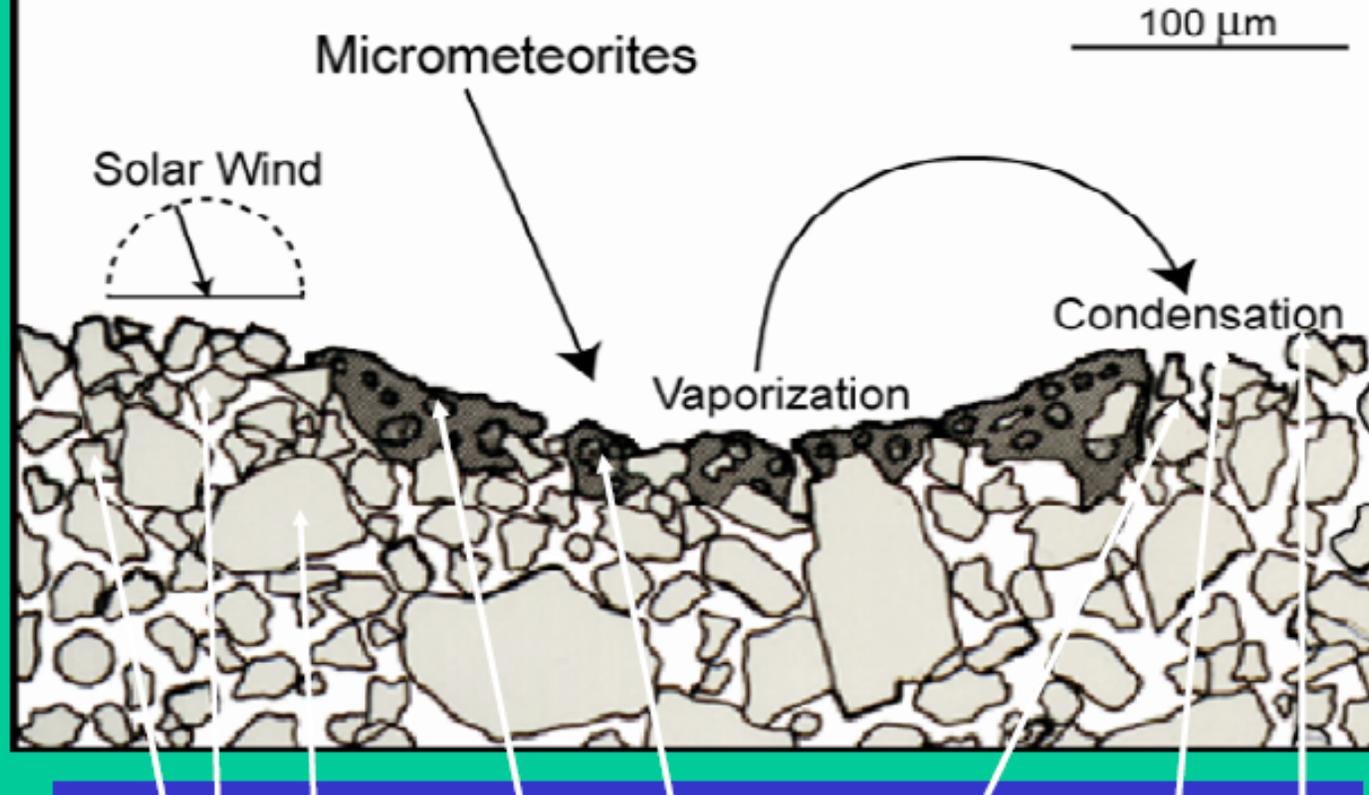
Lunar Mare Soil



**Regolith: broken up rock material; 'Soil': <1 cm portion of the Regolith
Dust: < 50 μm portion of the Soil**

ILC-2005: LA Taylor

Lunar Soil Formation



Comminution, Agglutination, & Vapor Deposition

**The major Weathering and Erosional agent on the Moon
is Meteorite and Micrometeorite Impact**

ILC-2005: LA Taylor

Production of JSC-1A



Mining of Lunar Soil Simulant at Flagstaff, Arizona

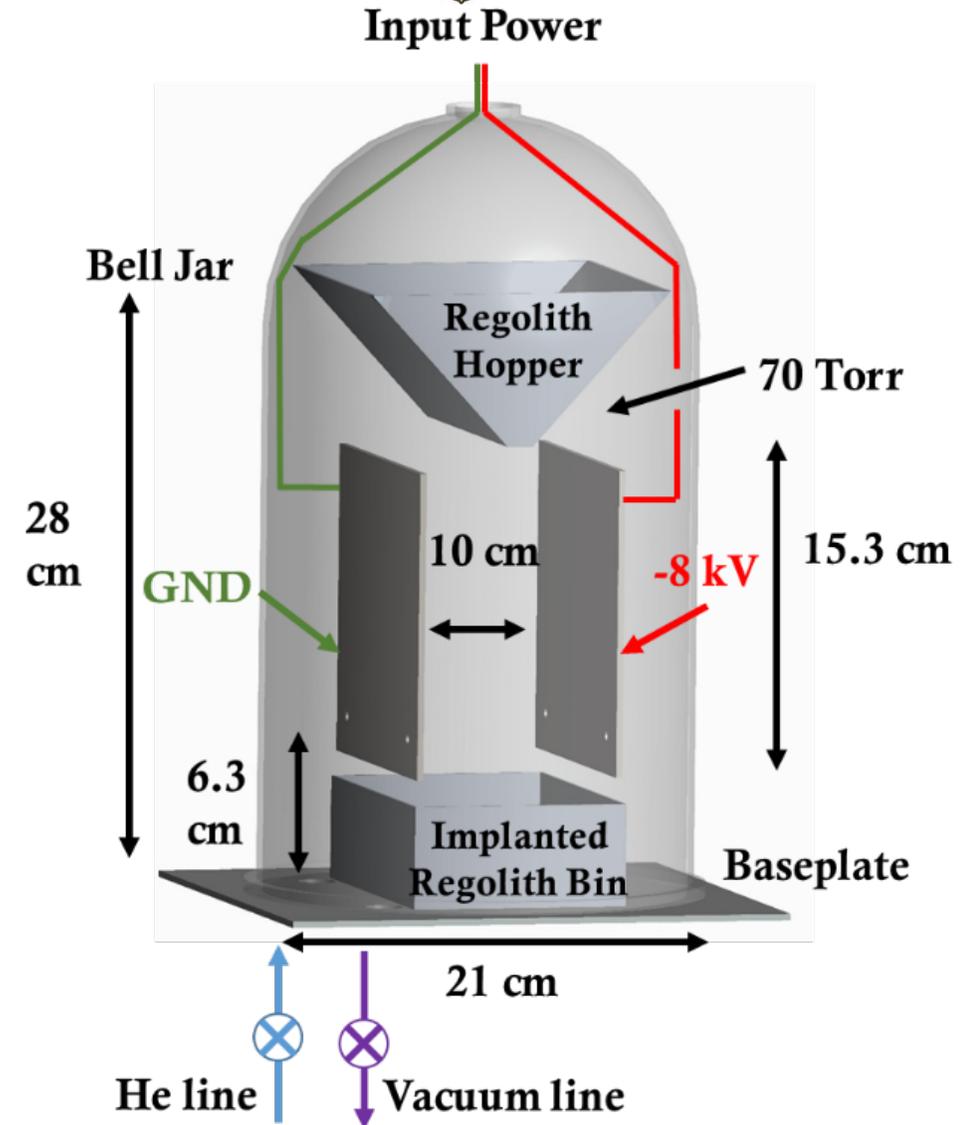
ILC-2005: LA Taylor



Regolith for the Heat Exchanger Must be Implanted with ^3He



- JSC-1A lunar regolith simulant will be used for the HEAT demonstration system
- JSC-1A does not contain any measurable amount of helium-3
- The simulant must be implanted with helium-3
- A dc glow discharge implantation system has been proposed to address this issue
- Simulant from a hopper will fall between parallel electrode plates inside of a vacuum chamber
 - Voltage between plates selected to simulate the energy helium ions have in the solar wind
 - Chamber pressure and distance between plates determined by the Paschen curve for helium



$$A' \ln Pd \exp(-B' Pd/V) = \ln(1 + (1/\gamma_{se}))$$

$$A' = 2.1 \text{ Pa}^{-1} \text{ m}^{-1}$$

$$B' = 57.75 \frac{V}{\text{Pa} \cdot \text{m}}$$

$$\gamma_{se} = 0.109$$